A1-F18AC-760-500 1 SEPTEMBER 1992 Change 3 - 1 November 2001

TECHNICAL MANUAL

ORGANIZATIONAL MAINTENANCE SYSTEM SCHEMATICS

TACTICAL ELECTRONIC WARFARE SYSTEMS

NAVY MODEL F/A-18A AND F/A-18B 161353 AND UP

N68936-01-D-0007

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NUMERICAL INDEX OF EFFECTIVE WORK PACKAGES/PAGES

List of Current Changes

Original 0 . . . 1 Sep 92 Change 2 1 Jan 96 Change 1 . . . 1 Dec 93 Change 3 1 Nov 01

Only those work packages/pages assigned to the manual are listed in this index. Insert Change 3, dated 1 November 2001. Dispose of superseded work packages/pages. Superseded classified work packages/pages shall be destroyed in accordance with applicable security regulations. If changed pages are issued to a work package, insert the changed pages in the applicable work package. The portion of text affected in a change or revision is indicated by change bars or the change symbol "R" in the outer margin of each column of text. Changes to illustrations are indicated by pointing hands, change bars, or MAJORCHANGE symbols. Changes to diagrams may be indicated by shaded borders.

WP Number	Title	WP Number	Title
Title		010 00	Interconnect Schematic -
Page A	Numerical Index of Effective Work Packages/Pages		Countermeasures Warning And Control System
TPDR-1	List of Technical Publications Deficiency Reports Incorporated	011 00	Power Interface Schematic -
001 00	Alphabetical Index		Countermeasures Warning and Control System
002 00 003 00 004 00	Introduction Locator - Interference Blanker System Functional Schematic - Interference	012 00	RF Detection and Conversion Schematic - Countermeasures
004 00	Blanker System Locator - Countermeasures	013 00	Warning and Control System Integration Schematic - Countermeasures Warning and
006 00	Dispensing System Functional Schematic - Countermeasures Dispensing System	014 00	Control System Video Processing and Control
007 00	Locator - Countermeasures Set		Schematic - Countermeasures
008 00	Functional Schematic - Countermeasures Set	015 00	Warning and Control System Controls Displays and Audio
009 00	Locator - Countermeasures Warning and Control System		Schematic - Countermeasures Warning and Control System

Total number of pages in this manual is 144, consisting of the following:

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C blank	3	1 - 7	2	12 blank .	3	007 00	
TPDR-1	3	8 blank	2	005 00		1 - 12	0
TPDR-2 blan	k 3	003 00		1 - 9	0		
001 00		1 - 7	3	10 blank .	0		
1	0	8	3				
2 blank	0						

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008 00	009 00	4 blank 1	014 00
1 0	1 - 16 0	012 00	1 - 5 1
2 blank 0	010 00	1 - 8 1	6 blank 1
008 01	1 - 9 2	013 00	015 00
1 - 4 0	10 blank 2	1 - 7 0	1 - 14 0
008 02	011 00	8 blank 0	
1 - 10 0	1 - 3 1		

TPDR-1 (TPDR-2 blank)

Change 3 - 1 November 2001

LIST OF TECHNICAL PUBLICATION DEFICIENCY REPORTS INCORPORATED

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

TACTICAL ELECTRONIC WARFARE SYSTEMS

This WP supersedes TPDR WP, dated 1 December 1993.

1. The TPDRs listed below have been incorporated in this issue.

IDENTIFICATION NUMBER/ QA SEQUENCE NUMBER	LOCATION
NONE	

ALPHABETICAL INDEX

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

TACTICAL ELECTRONIC WARFARE SYSTEMS

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Countermeasures Dispensing System	
Locator	005 00
System Schematic	006 00
Countermeasures Set	
Locator	007 00
System Schematic	008 00
Countermeasures Warning and Control System	
Controls, Displays, and Audio Schematic	015 00
Integration Schematic	013 00
Interconnect Schematic	010 00
Locator	009 00
Power Interface Schematic	011 00
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System Schematic	004 00
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Introduction	002 00
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INTRODUCTION

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

TACTICAL ELECTRONIC WARFARE SYSTEMS

This WP supersedes WP002 00, dated 1 September 1992.

PURPOSE.

2. This manual has system schematics to give information about the system and allow signal tracing through the system. The system schematics support on-aircraft maintenance of mechanical, pneudraulic, electrical, and electronic functions. These functions are integrated on the schematics for ease of troubleshooting a complete system.

3. REQUISITIONING AND DISTRIBUTION OF NAVAIR TECHNICAL PUBLICATIONS.

4. Procedures to be used by Naval Activities and other Department of Defense organizations requiring NAVAIR technical publications are defined in the NAVAL AIR SYSTEMS COMMAND TECHNICAL MANUAL PROGRAM manual, NAVAIR 00-25-100 and NAVAIRINST 5605.5, Distribution of aeronautic technical publications. To automatically receive future changes and revisions to NAVAIR technical manuals, an activity must be established on the Automatic Distribution Requirements List (ADRL) maintained by the Naval Air Technical Services Facility (NAVAIRTECHSERVFAC). To become established on the ADRL, notify your activity central technical publications librarian. If your activity does not have a library, you may establish your automatic distribution requirements by contacting the Commanding Officer, NAVAIRTECHSERVFAC, Attn:

Code 321, 700 Robbins Avenue, Philadelphia, PA 19111-5097. Annual reconfirmation of these requirements are necessary to remain on automatic distribution. Please use your NAVAIRTECHSERVFAC assigned account number whenever referring to automatic distribution requirements.

If additional or replacement copies of this manual are required with no attendant changes in the ADRL, they may be ordered by submitting a DD 1348 requisition directly to the Commanding Officer, Naval Aviation Supply Office, Naval Publication and Forms Directorate, 5801 Tabor Road, Philadelphia, PA 19120-5099.

5 CONTENT.

- Each system is supported by schematics and a component locator.
- 7. COMPONENT LOCATOR. The component locator shows aircraft location, nomenclature and reference designation number of each system component. The illustration shows the technicians view when possible.
- SCHEMATICS. Simplified schematics, and detailed schematics provide direct support for testing and troubleshooting. All schematics are shown with electrical power off, switches

in off position, and relays in deenergized position unless noted on schematic.

- Simplified Schematics. Simplified schematics consist primarily of blocks connected by single lines with limited use of symbols and pictorial drawings of units. These schematics simplify system functions as much as possible.
- 10. **Detailed Schematics.** Detailed schematics integrate applicable electrical, pneudraulic and mechanical functions of the system. Detailed schematics show component location, connector pin letters and numbers, in line connectors, test points, and enough data to trace signals through the components within the system. Operational information next to components provides more data as required.

11. SCHEMATIC HIGHLIGHTS.

12. For schematic highlights see figure 1.

13. MANUAL ISSUE DATE.

14. The date on the title page is the copy freeze date. No additions, deletions, or changes are made after the manual issue date except last minute safety of flight or required maintenance changes. Data collected after the manual issue date will be included in later changes or revisions of the manual.

15. EFFECTIVITIES.

16. Effectivity notes on manual title pages, work package title pages, and within a work package indicate the aircraft or software program to which the data applies. If no effectivity note appears on the work package title page, the work package has the same effectivity as shown on the manual title page. The effectivity notes may use:

NOTE

Aircraft with model designator F/A-18B are the same type and model as TF/A-18A.

- a. Type, model, and series
- b. Bureau number (tail number)
- c. Combination of type, model, series, and bureau numbers
 - d. Part number or serial number
 - e. Technical directive number
 - f. Configuration/identification number
- 17. The table below shows examples of effectivity notes and their meanings:

Effectivity Note Examples

Effectivity Note	Definition
160777 AND UP	Applicable to all F/A-18A, F/A-18B, F/A-18C and F/A-18D for bureau numbers listed.
F/A-18A, F/A-18B	Applicable to all F/A-18A and F/A-18B.
F/A- 18C, F/A-18D	Applicable to all F/A-18C and F/A-18D.
F/A-18A	Applicable to all F/A-18A, but not F/A-18B, F/A-18C and F/A-18D.

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Effectivity Note Examples (Continued)

_	_
F/A-18B	Applicable to all F/A-18B, but not F/A-18A, F/A-18C, and F/A-18D.
F/A-18C	Applicable to all F/A-18C, but not F/A-18A, F/A-18B, and F/A-18D.
F/A-18D	Applicable to all F/A-18D, but not F/A-18A, F/A-18B, and F/A-18C.
F/A-18A, F/A-18C	Applicable to all F/A-18A and F/A-18C, but not to F/A-18B and F/A-18D.
F/A-18B, F/A-18D	Applicable to all F/A-18B and F/A-18D, but not to F/A-18A and F/A-18C.
F/A-18A 160775, 160777 THRU 160782	Only applicable to some bureau numbers of F/A-18A. Not applicable to any F/A-18B, even if a F/A-18B bureau number is within the numbers listed.
F/A-18C 163427, 163430 THRU 163456	Only applicable to some bureau numbers of F/A-18C. Not applicable to any F/A-18D, even if a F/A-18D bureau number is within the numbers listed.
F/A-18B 160784 AND UP	Only applicable to some bureau numbers of F/A 18B. Not applicable to any F/A-18A, even if an F/A-18A bureau number is within the numbers listed.
F/A-18D 163434 THRU 163457	Only applicable to some bureau numbers of F/A-18D. Not applicable to any F/A-18C, even if a F/A-18C bureau number is within the numbers listed.
160775 THRU 160785 BEFORE F/A-18 AFC 772	Applicable to F/A- 18A and F/A-18B for bureau numbers listed, before modification by technical directive.
161213 AND UP; ALSO 160775 THRU 160785 AFTER F/A-18 AFC 772	Applicable to aircraft modified during production; also applicable when affected aircraft have been modified by technical directive.
160775 THRU 160785; WHEN NO. 2 CONTROL PANEL P/N XXXX-X IS INSTALLED	Applicable to F/A-18A and F/A 18B for bureau numbers listed if panel P/N XXXX-X is installed. (Configuration before AVC).
161213 AND UP; ALSO 160775 THRU 160785; WHEN NO. 2 CONTROL PANEL P/N XXXX-Y (AVC-102) IS INSTALLED	Applicable to aircraft modified during production; also applicable to aircraft components modified to the production configuration by technical directive. (Configuration after AVC).
P/N MBEU65101-9, MBEU65101-10 & MBEU65105-3	Applicable to assemblies which are interchangeable between aircraft.

Effectivity Note Examples (Continued)

ENGINE NO. 215101 THRU 215109	Applicable to assemblies which are interchangeable between aircraft, but configurations can not be identified by part number.
CONFIG/IDENT NUMBER 84A	The CONFIG/IDENT Number is the program load identification number which identifies the software program loaded in specific programmable units. Refer to A1-F18AC-SCM-000 for CONFIG/IDENT Number tables.

18 TECHNICAL DIRECTIVES.

19. Technical directives are documents which direct the accomplishment, and recording of a retrofit configuration or inspection to delivered aircraft, or aircraft components.

20. AIRFRAME CHANGE (AFC) AND AIR-BORNE TACTICAL SOFTWARE CHANGE

(ASC). Technical directives which change configuration of aircraft structure or equipment installation, i.e. AFC, will list aircraft bureau numbers in effectivity notes and show before and after the AFC. Technical directives which change configuration of operational flight programs (OFP), i.e. ASC, will list the OFP CONFIG/IDENT NUMBER in effectivity notes and show the latest two authorized OFP programs. See AFC and ASC effectivity examples in Effectivity Note Example Table.

21. AIRCRAFT COMPONENT CHANGES. Technical directives which change configuration of aircraft components, i.e. AAC, ACC, AVC, AYC, and PPC will list part numbers in the effectivities. See AVC effectivity examples in Effectivity Note Example table.

22. HISTORICAL RECORD/RECORD OF APPLICABLE TECHNICAL DIRECTIVES.

23. The technical directives affecting this manual are listed in the Record of Applicable Technical Directives of each affected work package. Because an ASC directs all aircraft be modified within 30 days, ASC's are not listed. When all affected aircraft are modified, the before configuration is removed from the manual, and the technical directive entry is removed from the Record of Applicable Technical Directives and entered in the Historical Record of Applicable Technical Records.

24. TECHNICAL PUBLICATIONS DEFICIENCY REPORT (TPDR).

25. The TPDR (OPNAV FORM 4790/66) is the form for reporting errors and suspected omissions in the technical manuals. Reporting procedures are in OPNAVINST 4790 2 SERIES

26. NAVY (AN) STANDARD/COMMON NAME NOMENCLATURE.

27. When an item has both Navy (AN) standard and common name nomenclature assigned, the common name nomenclature will be used in text and on illustrations. Full Navy (AN) standard nomenclature will be used in the Illustrated Parts Breakdown (IPB).

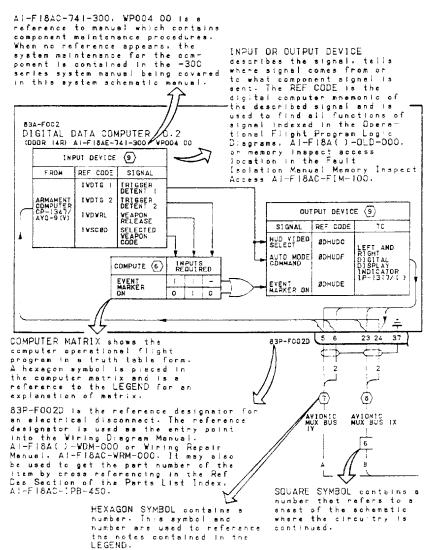
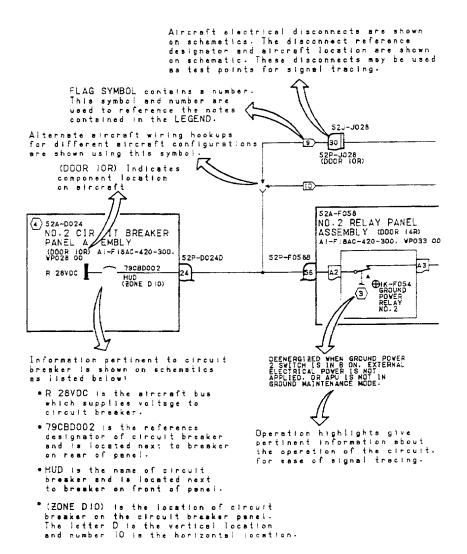


Figure 1. Schematic Highlights (Sheet 1)



The legend contains all notes pertinent to the schematic as listed below:

- NUMBER listed with no symbol is general information about the schematic.
- NONSTANDARD SYMBOLS appearing on schematic are shown or referenced with an explanation.
- ABBREVIATIONS appearing on schematic are shown or referenced with an explanation.
- HEXAGON SYMBOL refers to another schematic or manual for continuation of a circuit or an explanation of date contained on schematic.
- · FLAG SYMBOL indicates limited aircraft application.



1. CONTINUITY TESTS

- A. ALL AIRCRAFT WIRE NUMBERS, SPLICE POINTS, AND GROUND POINTS ARE SHOWN IN A1-F18A()-WDM-000.
- B. WHEN A LOW FUEL LEVEL CURRENT SWITCHING RELAY (IDENTIFIED BY \oplus) IS REMOVED FOR TROUBLESHOOTING. IDENTIFY RELAY AND SOCKET FOR CORRECT REINSTALLATION. DO NOT REPLACE LOW LEVEL CURRENT SWITCHING RELAY WITH ANY OTHER USED RELAY. IF RELAY IS DEFECTIVE REPLACE WITH NEW RELAY.
- C. DO NOT TEST LOW LEVEL DEVICES (SWITCHES/RELAY CONTACTS) FOR CONTINUITY WITH MULTIMETER ON RX1 SCALE. PIN TO PIN TESTS THAT DO NOT GO THROUGH SWITCHES RELAY CONTACTS MAY USE THE RX1 SCALE.
- D. WHEN TESTING CONTINUITY, TEST FOR:
 - (1) SHORTS TO GROUND.
 - (2) SHORTS BETWEEN SURROUNDING PINS ON CONNECTORS.
 - (3) SHORTS BETWEEN SHIELD AND CONDUCTORS.
 - (4) SHIELD CONTINUITY.
- 2. NONSTANDARD SYMBOLD:
 - ⊕ IDENTIFIES RELAY USED TO SWITCH TO LOW LEVEL CURRENT. SEE NOTE 1.
- GROUND POWER SWITCHING SCHEMATIC, A1-F18AC-420-500, WP005 00.
- POWER DISTRIBUTION SCHEMATIC, A1-F18AC-420-500, WP004 00.
- (5) EXPLANATION OF MATRIX:
 - A. COMPUTE COLUMN LISTS THE SIGNAL OUTPUT.
 - B. IMPUTS REQUIRED ARE USED TO DEVELOP THE SIGNAL OUTPUT.
 - C. THE SIGNAL OUTPUT IS READ HORIZONTALLY, EACH HORIZONTAL LINE IS AN INDEPENDENT SIGNAL OUTPUT.
 - D. INTERPRET MATRIX TABLE AS INDICATED:
 - (1) ONE (1) INDICATES THIS INPUT AS NAMED MUST BE THERE TO GET THE OUTPUT.
 - (2) ZERO (0) INDICATES THE INPUT AS NAMED MUST NOT BE THERE TO GET THE OUTPUT.
 - (3) DASH (-) INDICATED THE OUTPUT DOES NOT DEPEND ON THE INPUT.
- (6) AVIONIC MUX CHANNEL 1 SCHEMATIC, A1-F18AC-741-500, WP003 00.
- AVIONIC MUX CHANNEL 2 SCHEMATIC, A1-F18AC-741-500, WP003 00.
- FOR LOGIC DIAGRAMS RELATING TO REF CODE, REFER TO A1-F18AC-OLD-000. FOR MEMORY INSPECT ACCESS LOCATION RELATING TO REF CODE. REFER TO A1-F18AC-FIM-100.
- 9 F/A-18A.
 - 10 F/A-18B.

HILIGHT-(F50-3)-GRID

Subject

F/A-18 AFC

Page No.

2

1 Nov 01

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

LOCATOR

INTERFERENCE BLANKER SYSTEM

This WP supersedes WP003 00, dated 1 September 1992.

Reference Material

None

Alphabetical Index

Interference Blanker System Locator, Figure 1

Record of Applicable Technical Directives					
Type/ Number	Date	Title and ECP No.	Date Incorp.	Remarks	
F/A-18 AFC 50	16 Oct 84	Tactical Electronic Warfare Systems, ALR-67 Countermeasures, Modification of (ECP MDA-F/A-18-003R1 C1/C2/C3)	15 Jun 85	-	
F/A-18 AFC 253	-	U.S. Naval Reserves A+ Avionics Upgrade, Incorporation of (ECP MDA-F/A-18-0560R1)	1 Nov 01	-	

U.S. Marine Corps Reserves A+ Avionics

Upgrade, Incorporation of (ECP MDA-F/A-18-0583)

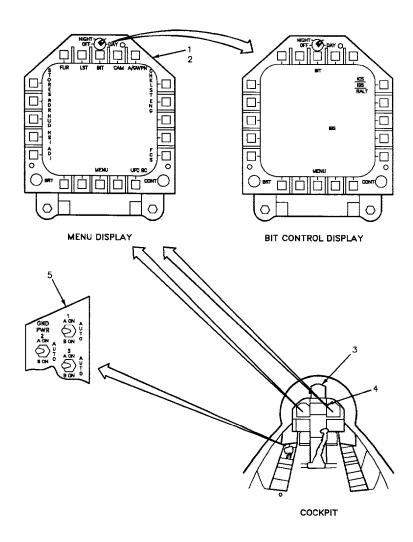
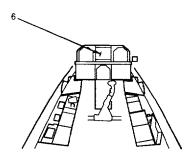


Figure 1. Interference Blanker System Locator (Sheet 1)



REAR COCKPIT

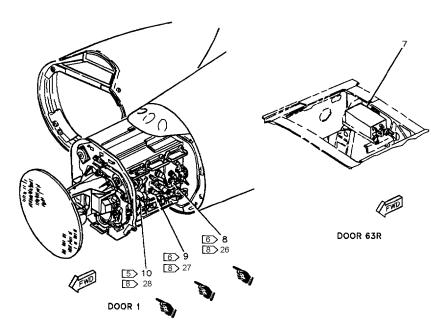
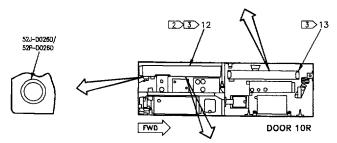


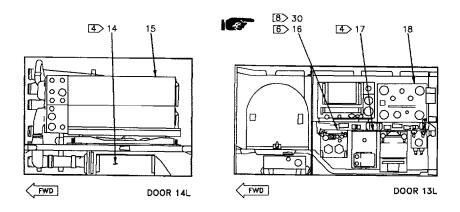
Figure 1. Interference Blanker System Locator (Sheet 2)

3 52A-D	024 N	NO. 2 CIRCUIT BREAKER PANEL ASSEMBL		
ZONE	REF DES	Nomenclature	BUS	
A11	82C9D002	CSC	R115VACØA	
A12	56080002	Blanker	R115VACØA	
B11	82CBD003	csc	R115VACØB	
C11	82CBD004	CSC	R115VACIØC	



ZONE	REF DES	NOMENCLATURE	BUS
> B3	82CBD005	csc	R28VDC
> 07	82CBD004	CSC	R115VACØC
> 08	82CBD003	CSC	R115VACØB
> 09	82CBD002	CSC	R115VACØA
C12	82CB0005	CSC	R28VDC
∑ D9	66CBD002	BLANKER	R115VACØA

Figure 1. Interference Blanker System Locator (Sheet 3)



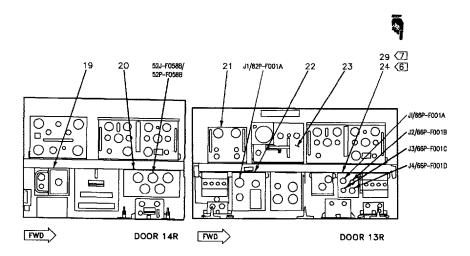


Figure 1. Interference Blanker System Locator (Sheet 4)

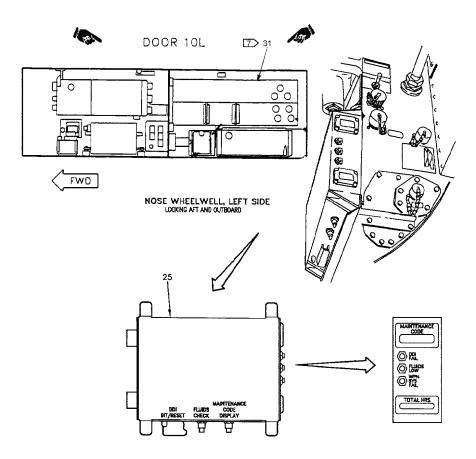


Figure 1. Interference Blanker System Locator (Sheet 5)

	NOMENCLATURE	INDEX NO.	REF DES
	COMMAND LAUNCH COMPUTER CP-1001()/AWG	23	61A-F010
6	COMPUTER POWER SUPPLY CP-1325/APG-65	9	60A-A505
	CONTROL-CONVERTER C-10382/A	22	82A-F001
4	COUNTERMEASURES COMPUTER CP-1293()/ALR-67(V)	17	62A-E006
	DIGITAL DATA COMPUTER NO. 1	18	83A-E001
	DIGITAL DISPLAY INDICATOR ID-2150/ASM-612	25	85A-G003
	ELECTRONIC EQUIPMENT CONTROL C-10380/ASQ	4	79A-J006
	GND PWR CONTROL PANEL ASSEMBLY	5	1A-H004
	HEAD-UP DISPLAY UNIT AN/AVQ-28	3	79A-J001
6	INTERFERENCE BLANKER MX-9965/A	24	66A-F001
7	INTERFERENCE BLANKER MX-11741/A	29	66A-F001
	LEFT DIGITAL DISPLAY INDICATOR IP-1317()	1	80A-H001
3	NO. 2 CIRCUIT BREAKER PANEL ASSEMBLY	13	52A-D024
	NO. 2 RELAY PANEL ASSEMBLY	20	52A-F058
	NO. 4 CIRCUIT BREAKER PANEL ASSEMBLY	12	52A-D026
7	NO. 7 CIRCUIT BREAKER/RELAY PANEL ASSEMBLY	31	52A-C057
8	RADAR DATA PROCESSOR CP-2062/APG-73	26	60A-A503
8	RADAR POWER SUPPLY PP-8318/APG-73	28	60A-A505
8	RADAR RECEIVER R-2484/APG-73	27	60A-A506
5	RADAR RECEIVER-EXCITER R-2089/APG-65	10	60A-A506
4	RADAR RECEIVER R-2055A/ALR-67(V)	14	62A-E009
	RADAR RECEIVER-TRANSMITTER RT-1028/APN-202	11	72A-A002
6	RADAR TARGET DATA PROCESSOR CP-1326/APG-65	8	60A-A503

Figure 1. Interference Blanker System Locator (Sheet 6)

	NOMENCLATURE	INDEX NO.	REF DES	
	NOMENCLATURE	INDEX NO.	REF DES	
	REAR ELECTRONIC EOUIPMENT CONTROL C-10380 ASO	6	76A-L028	
	RECEIVER-TRANSMITTER RT-1015()/APN-194(V)	7	67A-T001	
	RECEIVER-TRANSMITTER RT-1079()/ALQ-126	15	64A-E001	
6	RECEIVER-TRANSMITTER RT-1157()/APX-100(V)	16	78A-E001	
	RECEIVER-TRANSMITTER RT-1159/A	21	69A-F001	
8	RECEIVER-TRANSMITTER RT-1763/APX-111(V)	30	78A-E016	
	RIGHT DIGITAL DISPLAY INDICATOR IP-1317()	2	80A-J002	
	SIGNAL DATA RECORDER RO-508/ASM-612	19	85A-F001	

LEGEND

	1.	AIRCRAFT CONNECTOR LOCATIONS ARE SHOWN IN A1-F18A()-WDM-000.
	2	161353 THRU 161359.
	3	161360 AND UP.
	4	161702 AND UP.
i	5	$161353\ \mathrm{THRU}\ 161583;\ \mathrm{ALSO}\ 161702\ \mathrm{THRU}\ 163175\ \mathrm{BEFORE}\ \mathrm{F/A-18}\ \mathrm{AFC}\ 50,\ \mathrm{INTERFACE}\ \mathrm{WITH}\ \mathrm{INTERFERENCE}\ \mathrm{BLANKER}\ \mathrm{SYSTEM}\ \mathrm{EXISTS}.$
	6	161353 AND UP BEFORE F/A-18 AFC 253 OR F/A-18 AFC 292.
	7	162394 THRU 163175 AFTER F/A-18 AFC 253 OR F/A-18 AFC 292.
	8	162394 THRU 163175 AFTER F/A-18 AFC 292.

Figure 1. Interference Blanker System Locator (Sheet 7)

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

SCHEMATIC - FUNCTIONAL

INTERFERENCE BLANKER SYSTEM

This WP supersedes WP004 00, dated 1 September 1992.

Reference Material

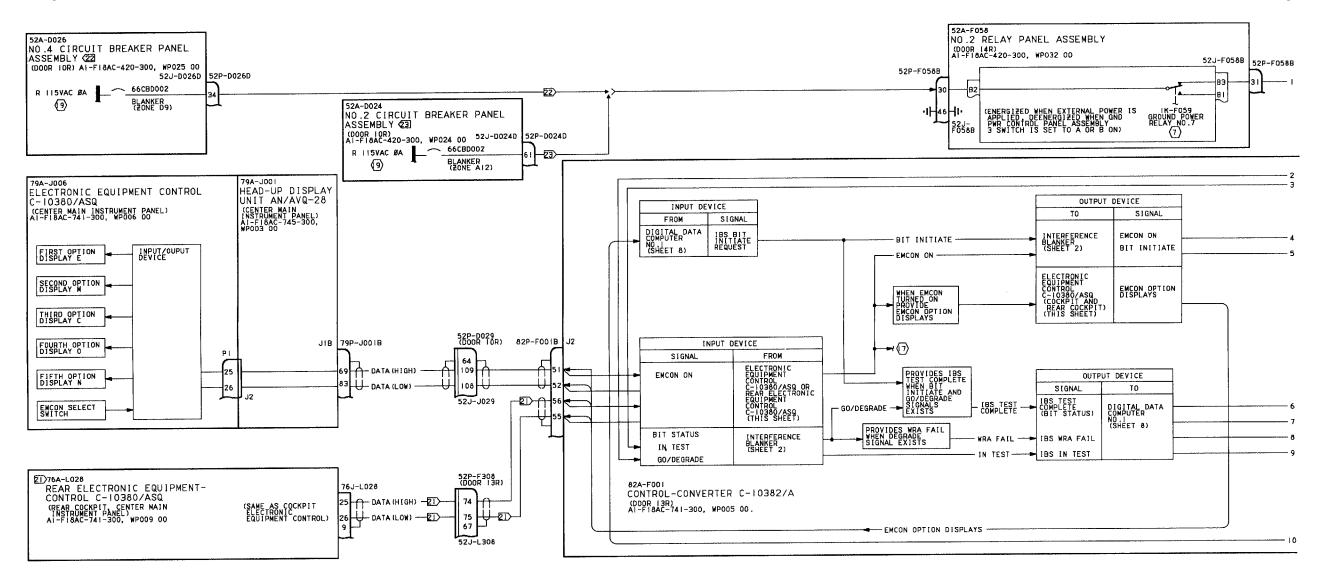
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Alphabetical Index

Subject	Page No.
Interference Blanker System Functional Schematic, Figure 1	2

Record of Applicable Technical Directives

Type/ Number	Date	Title and ECP No.	Date Incorp.	Remarks
F/A-18 AFC 50	16 Oct 84	Tactical Electronic Warfare System, ALR-67 Countermeasures, Modification of (ECP MDA- F/A-18-003R1 C1/C2/C3)	15 Jun 85	-
F/A-18 AFC 158	-	Correction of AN/ALR-67 and AN/ALQ-126B Wiring (ECP RAM EC NORIS-22-90)	1 Sep 92	-
F/A-18 AFC 253	-	U.S. Naval Reserves A+ Avionics Upgrade, Incorporation of (ECP MDA-F/A-18-0560R1	1 Nov 01	-
F/A-18 AFC 292	-	U.S. Marine Corps Reserves A+ Avionics Upgrade, Incorporation of (ECP MDA-F/A-18-0583)	1 Nov 01	-



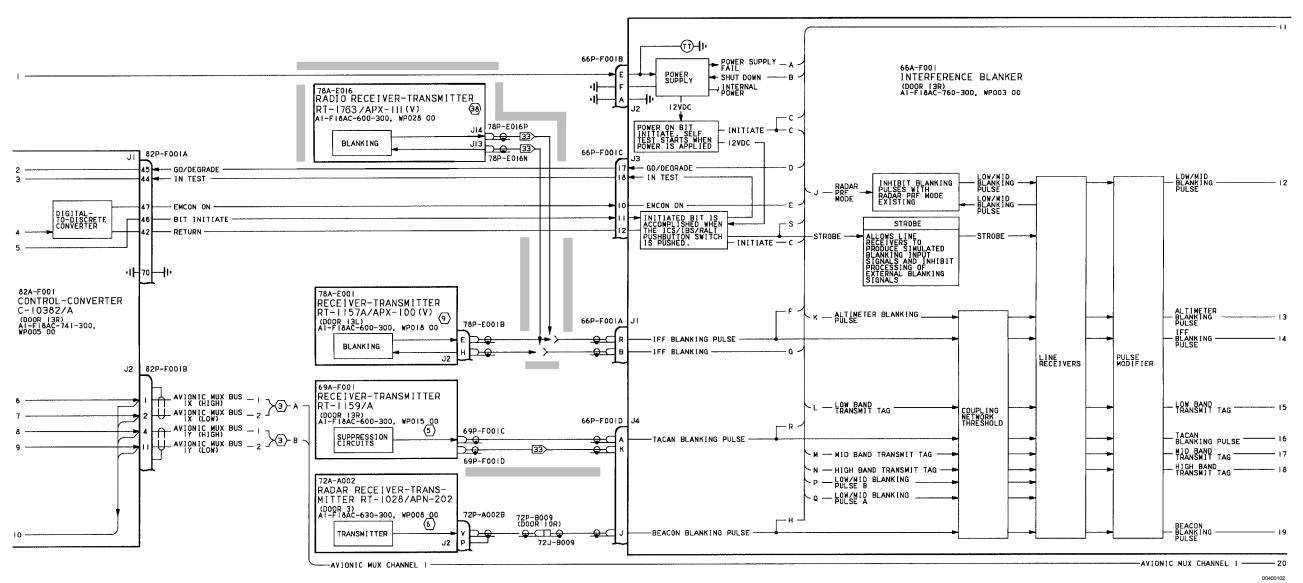
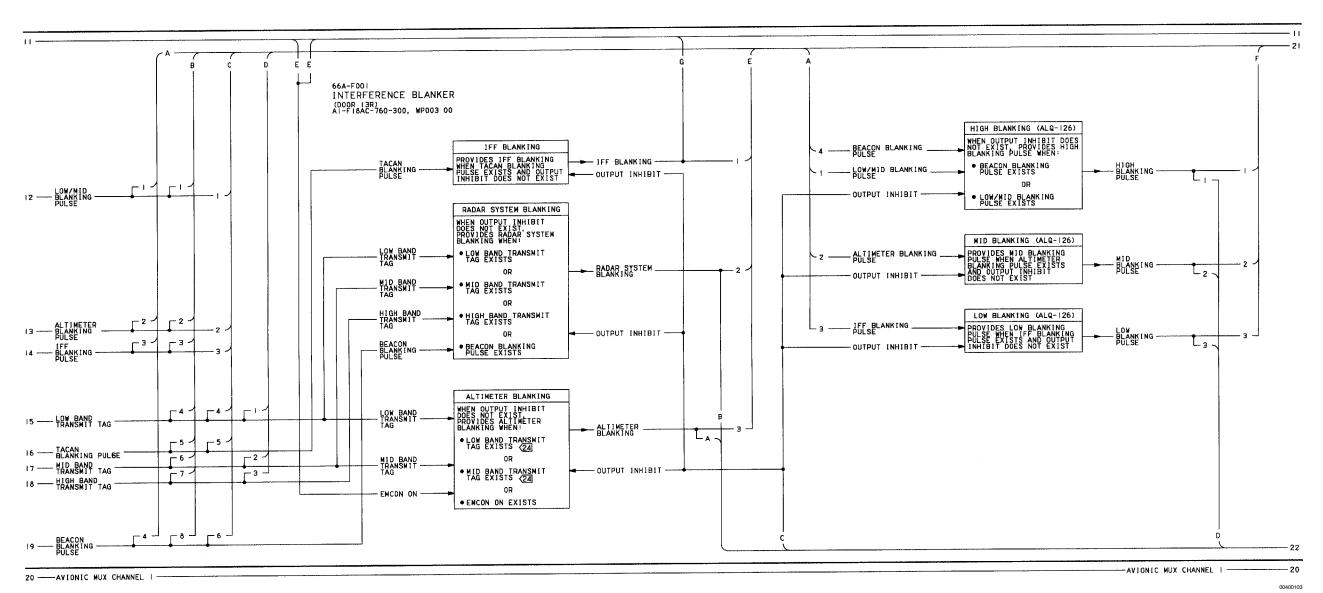
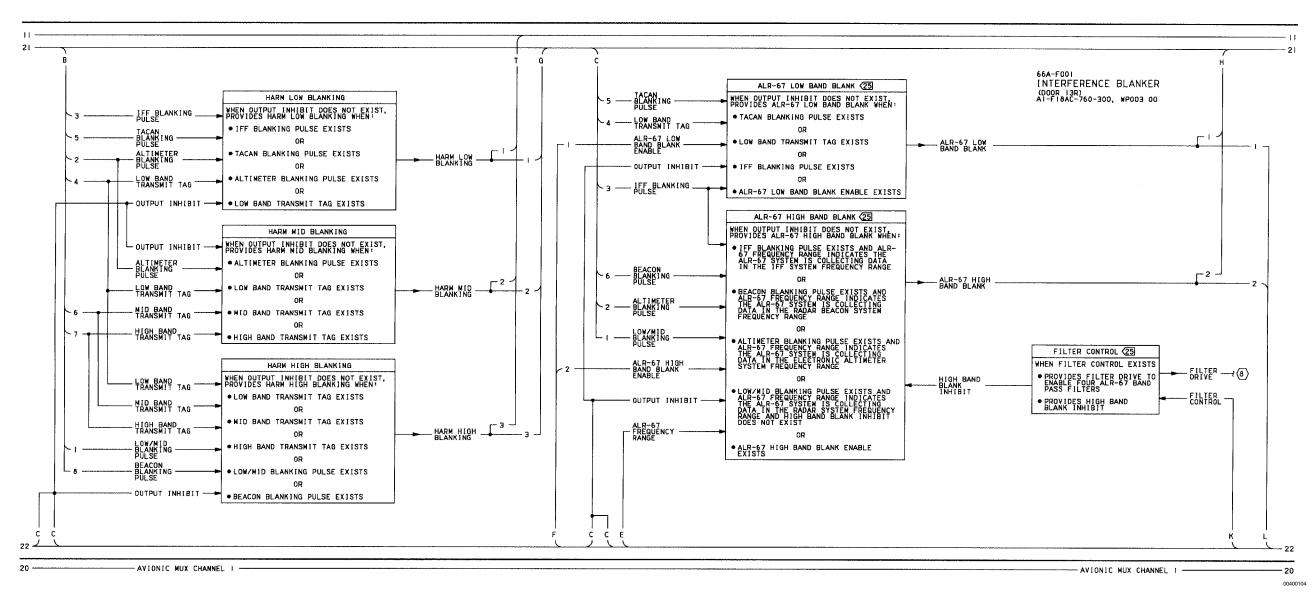
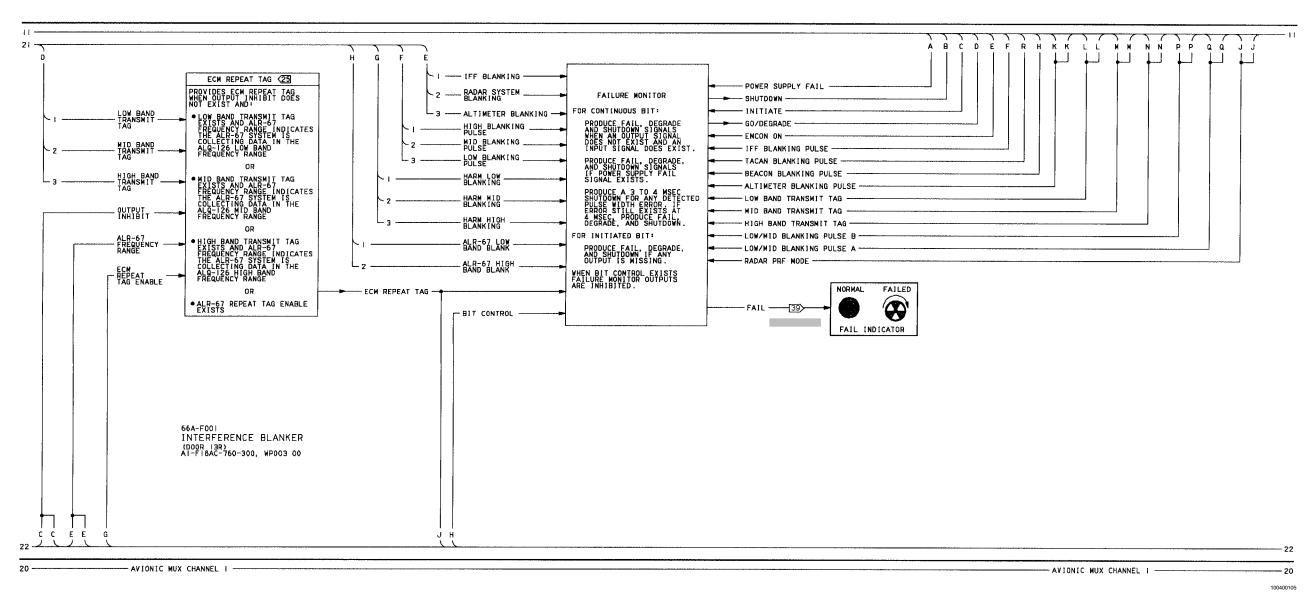


Figure 1.

Figure 1. Interference Blanker System Functional Schematic (Sheet 2)







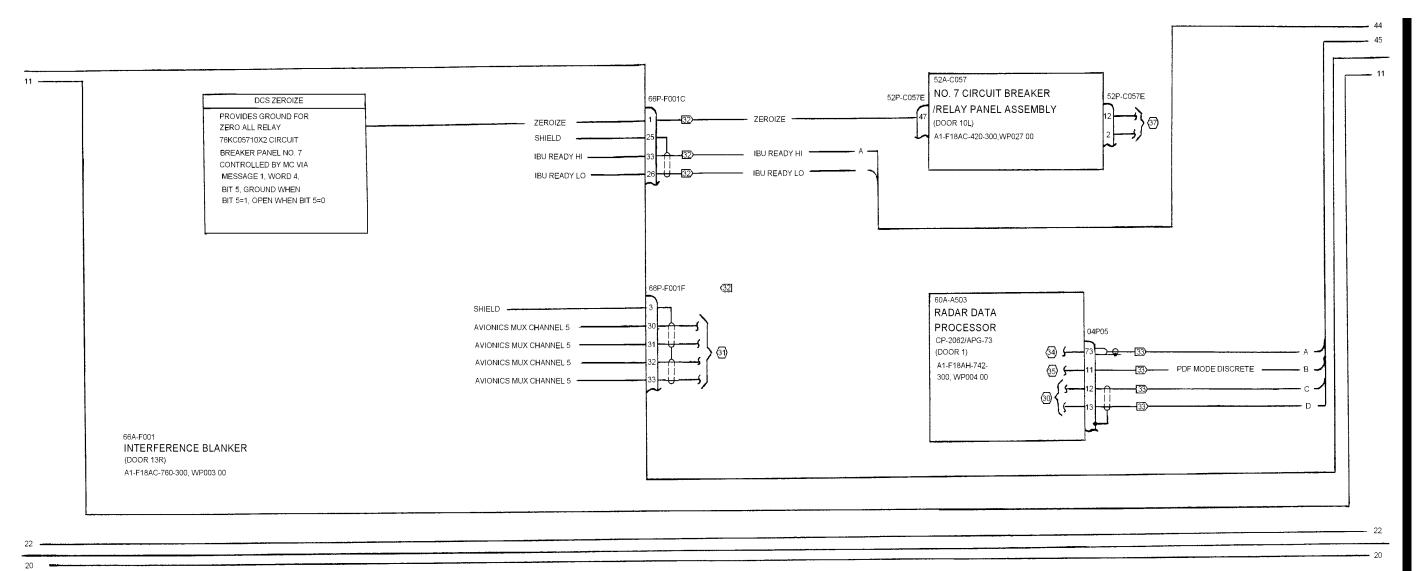


Figure 1.

Figure 1. Interference Blanker System Functional Schematic (Sheet 6)

0040010

Figure 1.

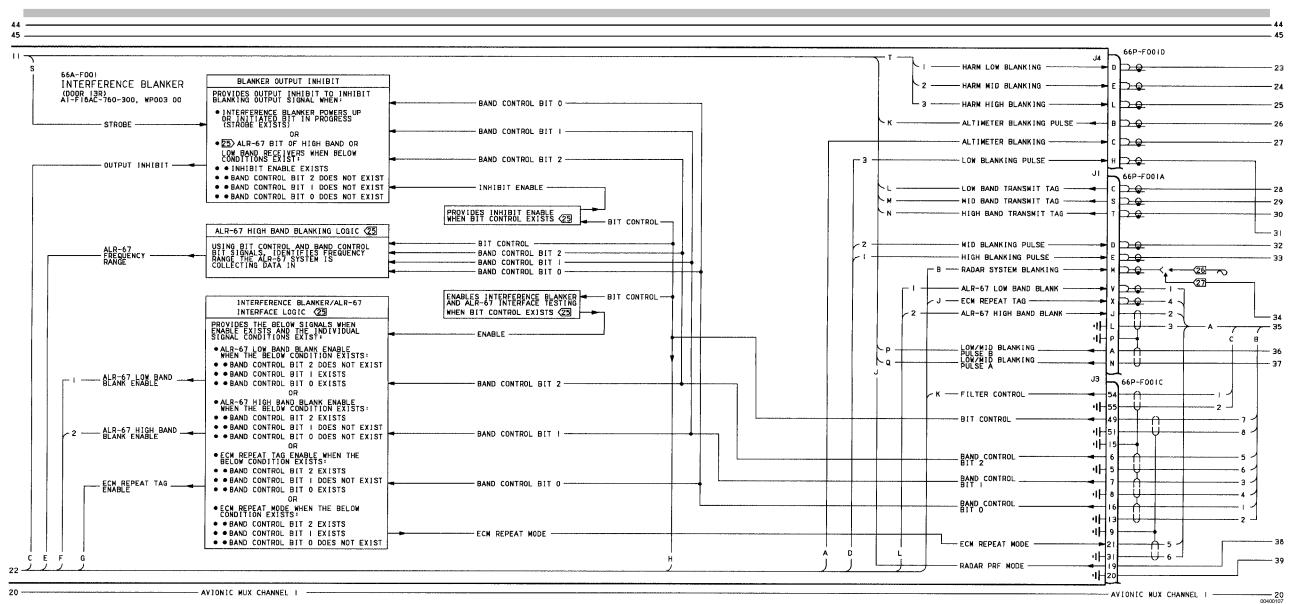


Figure 1. Interference Blanker System Functional Schematic (Sheet 7)

Figure 1.

Change 3

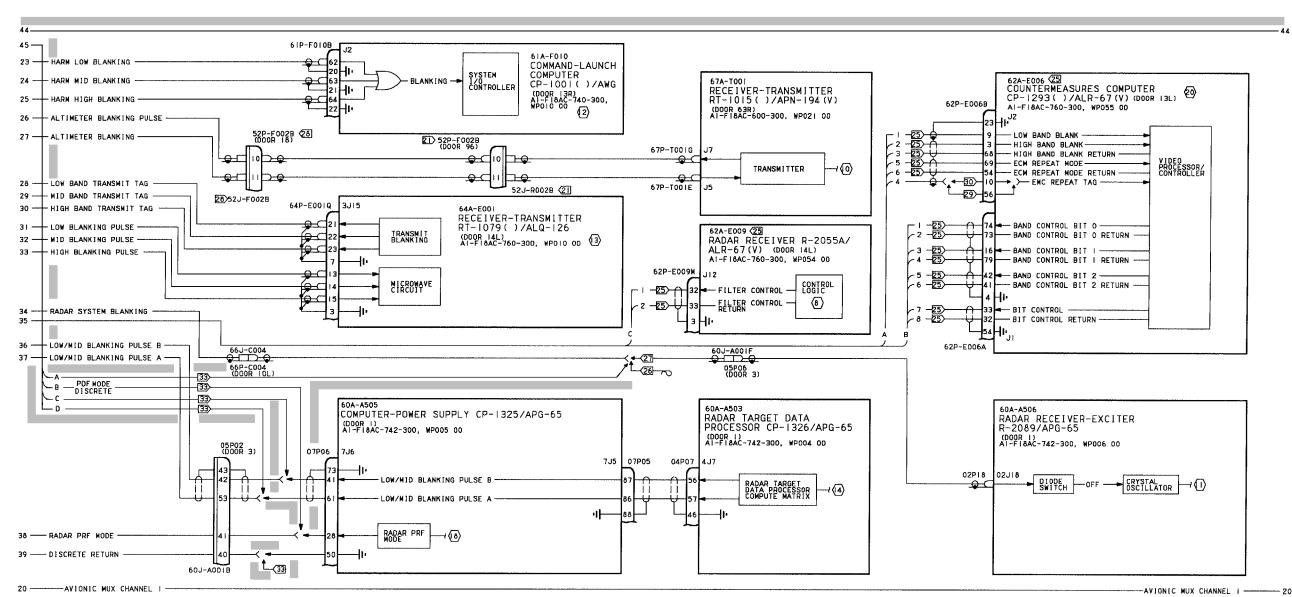


Figure 1. Interference Blanker System Functional Schematic (Sheet 8)

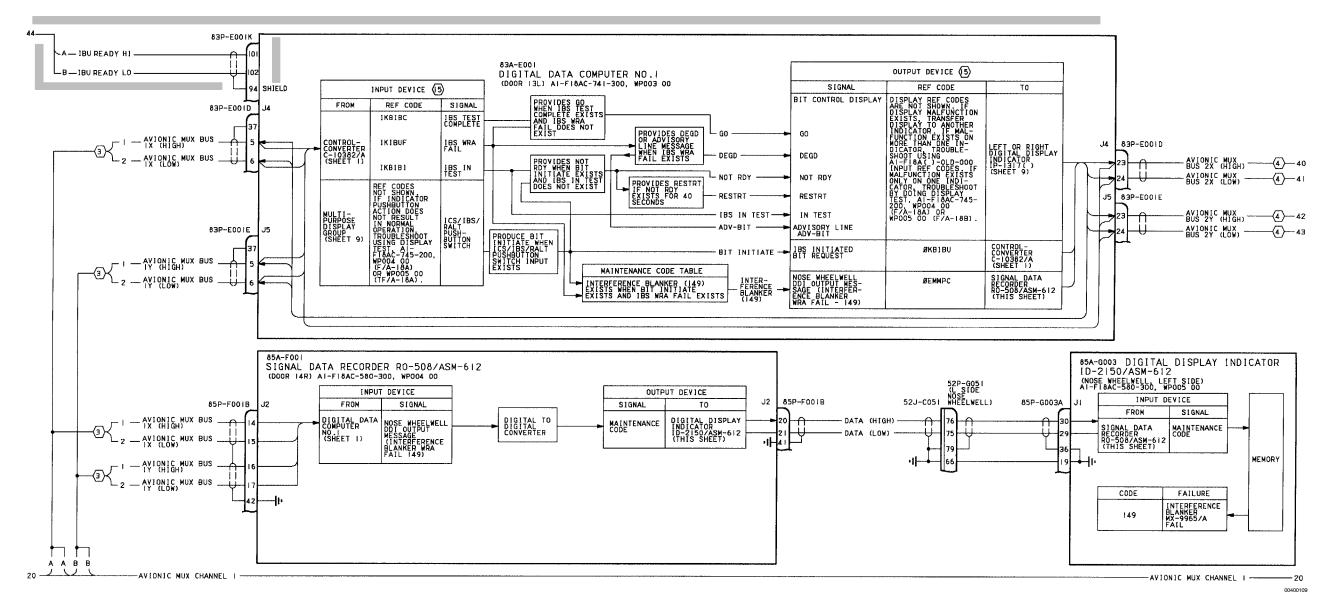
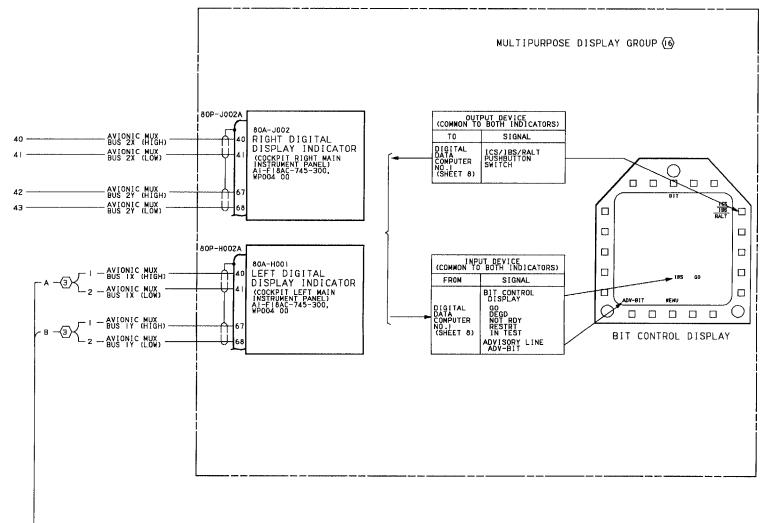


Figure 1.

Figure 1. Interference Blanker System Functional Schematic (Sheet 9)



LEGEND

- I. CONTINUITY TESTS:
 - A. ALL AIRCRAFT WIRE NUMBERS, SPLICE POINTS, AND GROUND POINTS ARE SHOWN IN AI-F18A()-WDM-000.
 - B. WHEN A LOW LEVEL CURRENT SWITCHING RELOY IN THE METERS OF THE SWITCHING RELOY CURRENT SWITCHING RELOY CURRECT REINSTALLATION DO NOT REPLACE LOW LEVEL CURRENT SWITCHING RELAY METERS OF THE RELAY IS DEFECTIVE, REPLACE WITH NEW RELAY IS DEFECTIVE,
 - C. DO NOT TEST TOW LEVEL DEVICES (SWITCHES/ RELAY CONTACTS) FOR CONTINUITY WITH HULTI METER ON RY SCALE PIN TO PIN TESTS THAT OF NOT GO THOUGH, SWITCHES/RELAY CONTACTS MAY USE THE RY SCALE.
 - D. WHEN TESTING CONTINUITY, TEST FOR: (I) SHORTS TO GROUND.
 - (2) SHORTS BETWEEN SURROUNDING PINS ON CONNECTORS.
 - (3) SHORTS BETWEEN SHIELD AND CONDUCTORS.
 - (4) SHIELD CONTINUITY.
- NONSTANDARD SYMBOLS:
- ## IDENTIFIES RELAY USED TO SWITCH LOW LEVEL CURRENT. SEE NOTE | .
- 3 AVIONIC MUX CHANNEL I SCHEMATIC, AI-FI8AC-741-
- 4 AVIONIC MUX CHANNEL 2 SCHEMATIC, AI-FI8AC-741-500, WP005 00.
- (5) TACAN SYSTEM FUNCTIONAL SCHEMATIC, AI-F18AC-600-500, WP016 00.
- (6) RADAR BEACON SYSTEM FUNCTIONAL SCHEMATIC, A1-F18AC-630-500, WP006 00.
- T GROUND POWER SWITCHING SCHEMATIC, AI-FIBAC-420-500, WP005 00. (B) RF DETECTION AND CONVERSION SCHEMATIC, WP012 00
- 9) IFF MODES 1 2 3/A AND C FUNCTIONAL SCHEMATIC, NP019 00.
- ELECTRONIC ALTIMETER SYSTEM FUNCTIONAL SCHEMATIC, AI-FIBAC-600-500, WP023 00.
- RADAR RECEIVING/AGC SIMPLIFIED SCHEMATIC, A1-F18AC-742-500, MP031 00.

- (2) AGM-88 HARM AVIONIC INTERFACE SCHEMATIC, AI-FIBAC-740-520, WP059 00.
- RECEIVER-TRANSMITTER RT-1079()/ALQ-126 FUNCTIONAL SCHEMATIC, WP008 00.
- TRANSMITTER DRIVE SCHEMATIC, AI-FIBAC-742-500, WF009 00.
- (5) FOR LOGIC DIAGRAMS RELATING TO REF CODE, REFER TO A1-F18AC -01D-0000. FOR MEMORY INSPECT ACCESS LOCATION RELATING TO REF CODE, REFER TO A1-F18AC-FIM-100.
- (6) SEE MULTIPURPOSE DISPLAY GROUP INTERCONNECT SCHEMATIC, AI-FIBAC-745-500, WP004 00.
- (17) EMCON SCHEMATIC, A1-F18AC-741-500, WP012 00
- B LAUNCH INITIATE AND MISSILE COMPATABLE PRF
- POWER DISTRIBUTION SCHEMATIC, AI-FIBAC-420-500.
- (20) VIDEO PROCESSING AND CONTROL SCHEMATIC, WP014 00.
- 21) F/A-18B
- 22 161353 THRU 161359.
- 23 161360 AND UP.
- ALTIMETER BLANKING PRODUCED WHEN INTERFERENCE ADSAUGZ8-4 9NSTALLED.
- 25 161702 AND UP
- 26) 163427 AND UP: ALSO 161702 THRU 163175 AFTER F/A-18 AFC 050.
- 27) 161353 THRU 161528; ALSO 161702 THRU 163175 BEFORE F/A-18 AFC 050.
- 28) F/A-18A.
- 29 161737 AND UP; ALSO 161702 THRU 161736 AFTER F/A-18 AFC 158.
- 30 161702 THRU 161736 BEFORE F/A-18 AFC 158
- 3 AVIONIC MUX CHANNEL 5 SCHEMATIC, AI-FIBAC-741-500, WP018 00.
- DE F/A-18A 162394 THRU 163175 AFTER F/A-18 AFC 253 OR F/A-18 AFC 292.
- \$3 F/A-18A 162394 THRU 163175 AFTER F/A-18 AFC 292.
- 34 RECIEVE PATH FUNCTIONAL SCHEMATIC, AI-FI8AH-742-500,
- 35) PRF SELECTION AND DISPLAY SCHEMATIC, AI-FI8AH-742-500,
- (36) RF PATH FUNCTIONAL SCHEMATIC, AI-FIBAH-742-500,
- WP009 00.
- (37) ZEROIZE CIRCUIT, AI-F18AC 600 500, WP004 00.
- (38) IFF MODES 1,2,3A, AND C FUNCTIONAL SCHEMATIC. AI-FIBAC-600-500, WP019 00.
- 99 F/A-18A 161353 AND UP BEFORE F/A-18 AFC 253 OR F/A-18 AFC 292.

20 ~

SYSTEM SCHEMATICS

LOCATOR

COUNTERMEASURES DISPENSING SYSTEM

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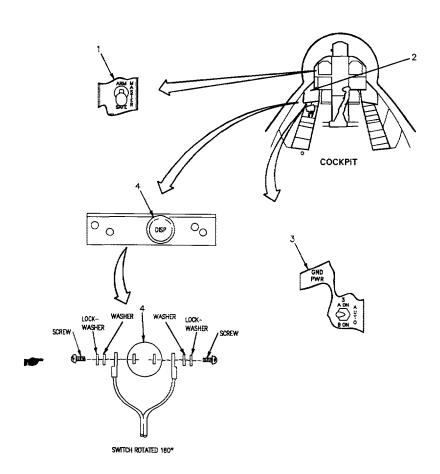
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Record of Applicable Technical Directives

None



18AC-760-50-(4-1)14-CATI

Figure 1. Countermeasures Dispensing System Locator (Sheet 1)

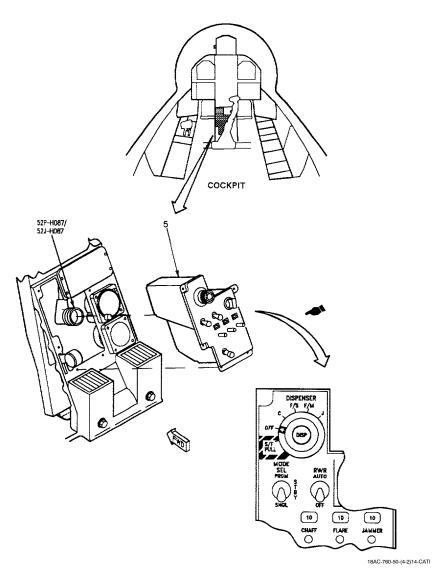
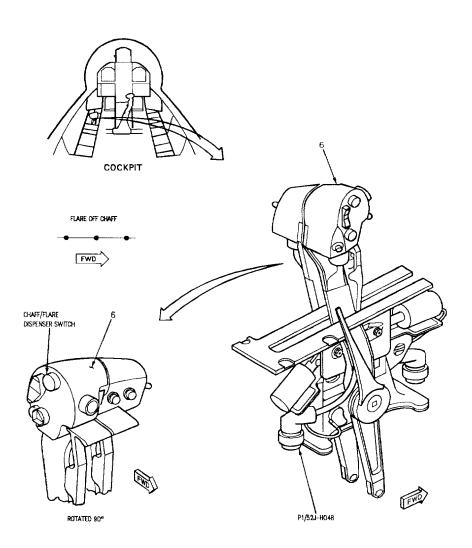
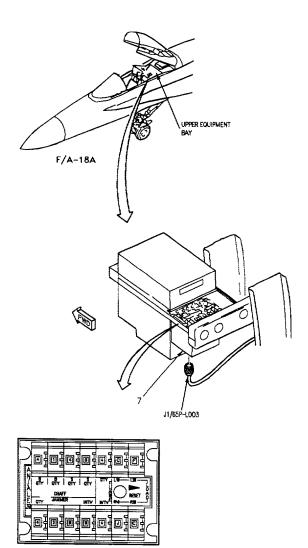


Figure 1. Countermeasures Dispensing System Locator (Sheet 2)



18AC-760-50-(4-3)13-CATI

Figure 1. Countermeasures Dispensing System Locator (Sheet 3)



18AC-760-50-(4-4)13-CATI

Figure 1. Countermeasures Dispensing System Locator (Sheet 4)

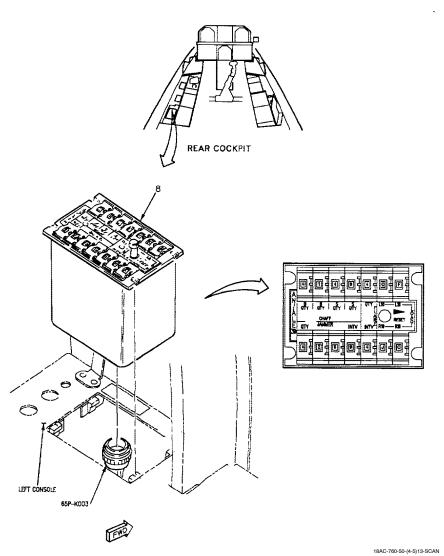


Figure 1. Countermeasures Dispensing System Locator (Sheet 5)

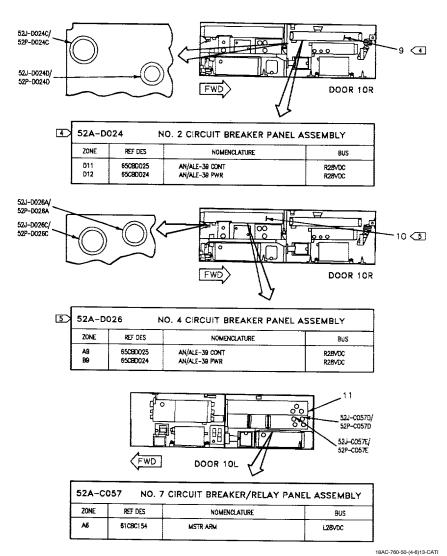


Figure 1. Countermeasures Dispensing System Locator (Sheet 6)

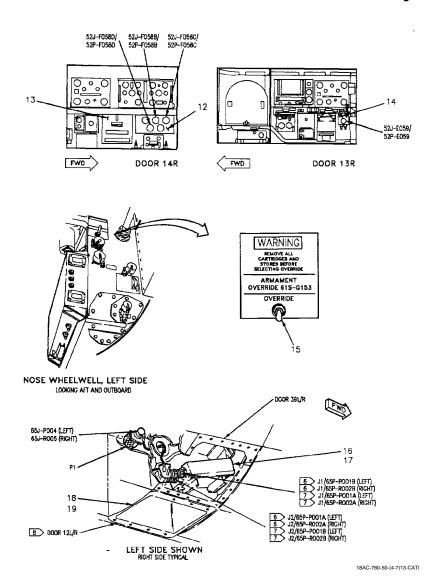


Figure 1. Countermeasures Dispensing System Locator (Sheet 7)

	NOMENCLATURE	INDEX NO.	REF DES
	ARMAMENT COMPUTER CP-1342/AYQ-9(V)	13	61A-F001
	ARMAMENT OVERRIDE SWITCH	15	61S-G153
	ECM CONTROL PANEL ASSEMBLY	5	52A-H087
	ECM DISP SWITCH	4	65A-H027
	GND PWR CONTROL PANEL ASSEMBLY	3	1A-H004
	LDG GEAR CONTROL	2	12A-H008
	LEFT DISPENSER HOUSING MX-7721/ALE-29A	18	65A-P004
	LEFT ELECTRICAL SWITCHING UNIT SA-1874/ALE-39	17	65A-P001
	MASTER ARM CONTROL PANEL ASSEMBLY	1	52A-H075
4	NO. 2 CIRCUIT BREAKER PANEL ASSEMBLY	9	52A-D024
	NO. 2 RELAY PANEL ASSEMBLY	12	52A-F058
	NO. 3 RELAY PANEL ASSEMBLY	14	52A-E059
5	NO. 4 CIRCUIT BREAKER PANEL ASSEMBLY	10	52A-D026
	NO. 7 CIRCUIT BREAKER/RELAY PANEL ASSEMBLY	11	52A-C057
3	PROGRAMMER MX-9254/ALE 39	7	65A-L003
2	PROGRAMMER MX-9254/ALE-39	8	65A-K003
	RIGHT DISPENSER HOUSING MX-7721/ALE-29A	19	65A-R005
	RIGHT ELECTRICAL SWITCHING UNIT SA-1874/ALE-39	16	65A-R002
	RIGHT THROTTLE GRIP	6	52A-H048

LEGEND

1.	AIRCRAFT CONNECTOR LOCATIONS ARE SHOWN IN A1-F18A()-WDM-000.
2	F/A-18B
3	F/A-18A.
4	161353 THRU 161359.
5	161360 AND UP.
6	161353 THRU 161521.
7	161522 AND UP.
8	162826 AND UP.

Figure 1. Countermeasures Dispensing System Locator (Sheet 8)

SYSTEM SCHEMATICS

SCHEMATIC - FUNCTIONAL

COUNTERMEASURES DISPENSING SYSTEM

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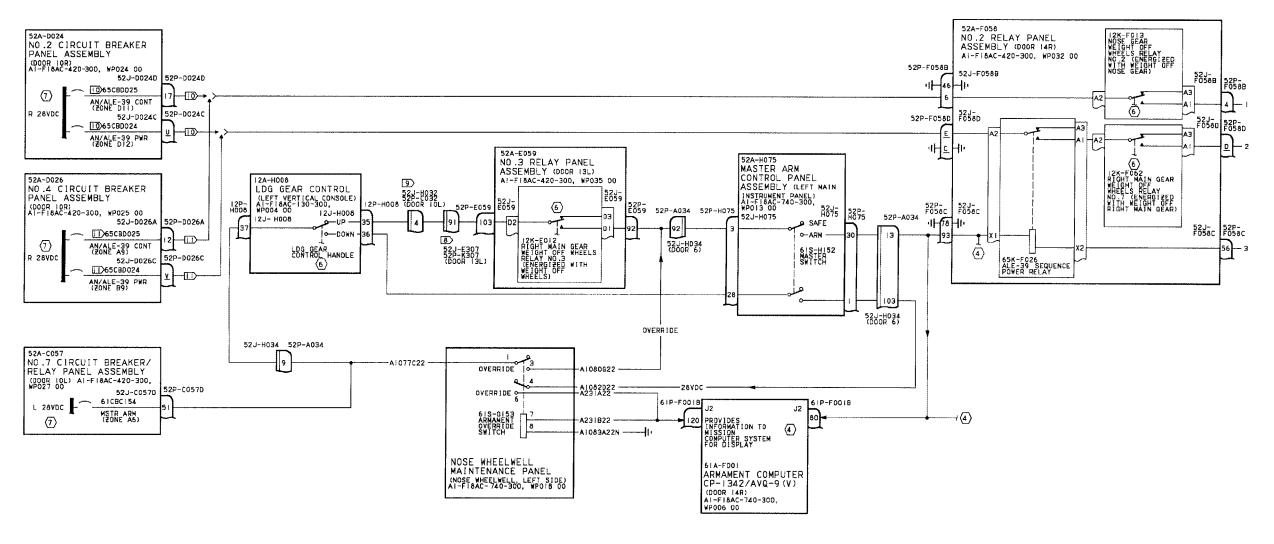
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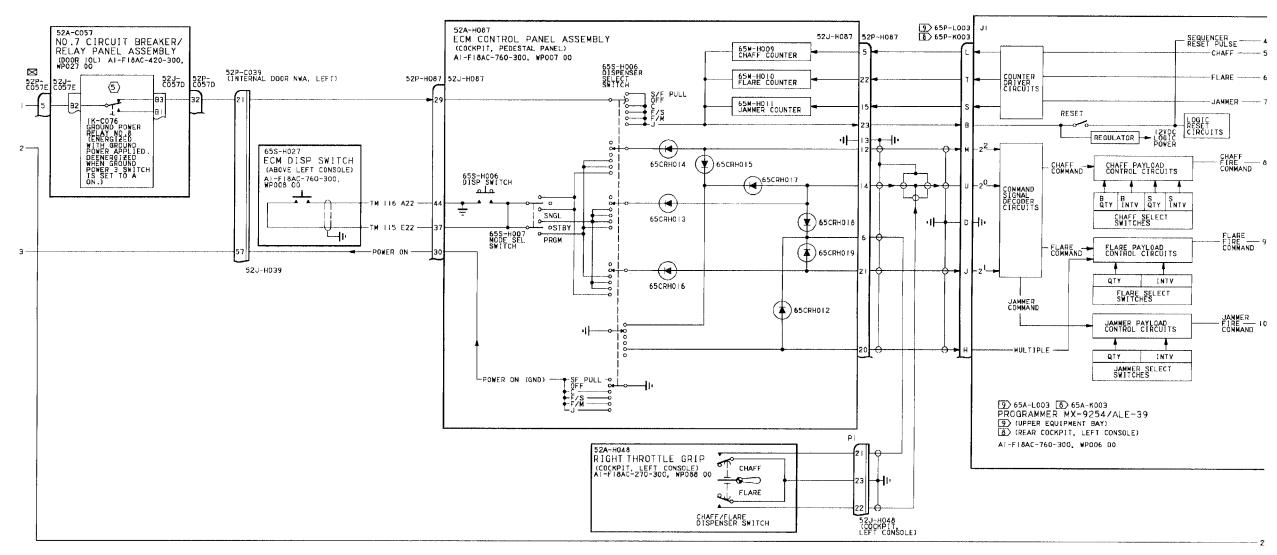
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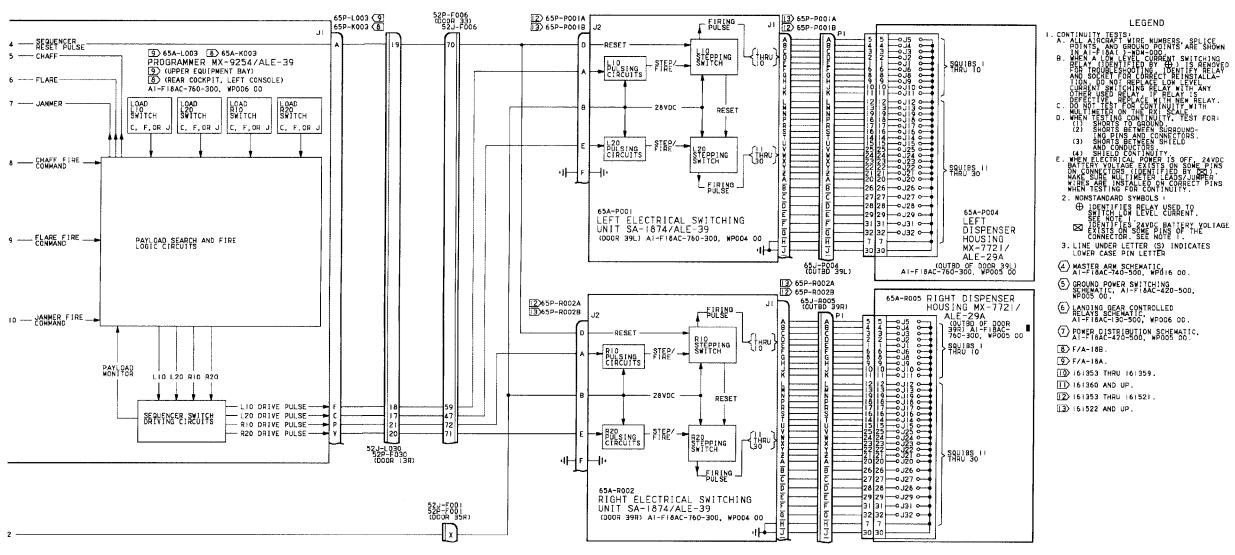
Subject	Page No.
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Record of Applicable Technical Directives

None







1 September 1992

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

LOCATOR

COUNTERMEASURES SET

Reference Material

None

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Record of Applicable Technical Directives

Type/ Number	Date	Title and ECP No.	Date Incorp.	Remarks
F/A-18 AFC 50	16 Oct 84	Tactical Electronic Warfare Systems, ALR-67 Countermeasures, Modification of (ECP MDA-F/A-18-003R1 C1/C2/C3)	15 Jun 85	-

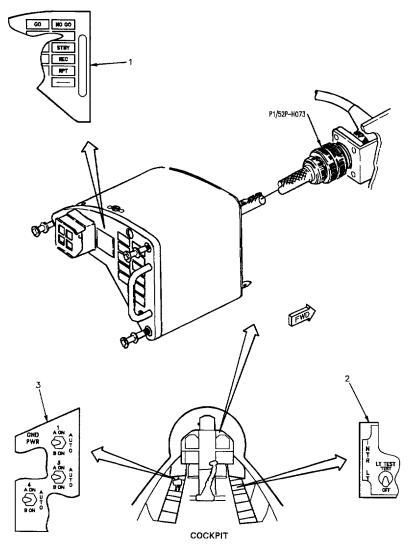
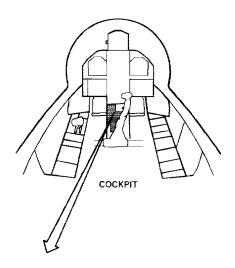


Figure 1. Countermeasures Set Locator (Sheet 1)

18AC-760-50-(7-1)13-SCAN



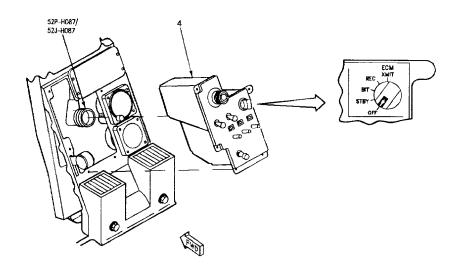


Figure 1. Countermeasures Set Locator (Sheet 2)

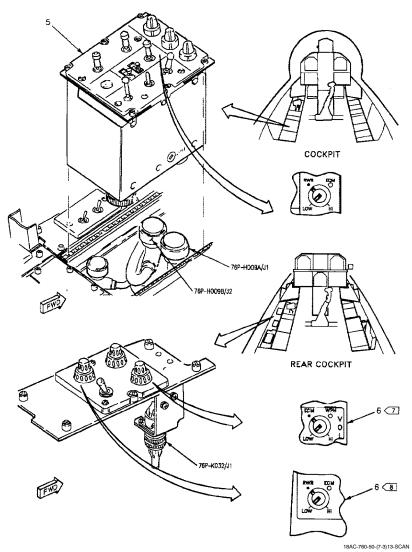
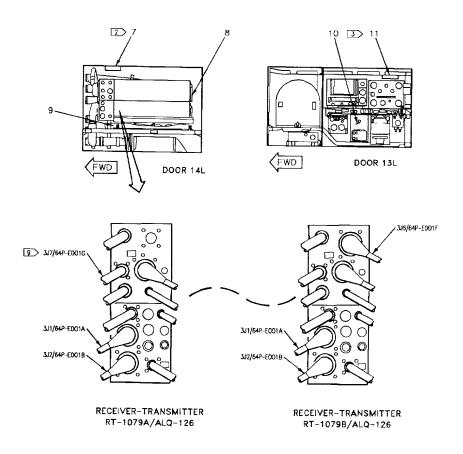


Figure 1. Countermeasures Set Locator (Sheet 3)



18AC-760-50-(7-4)13-CATI

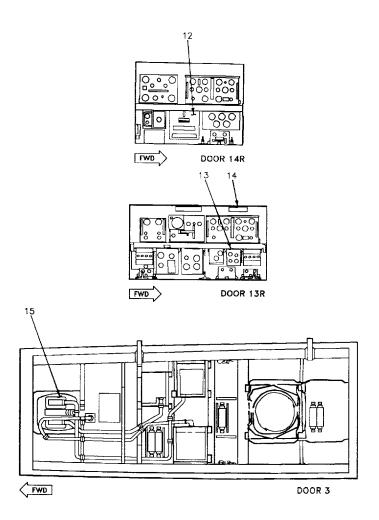
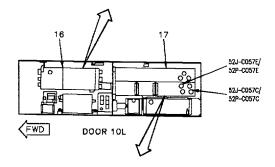
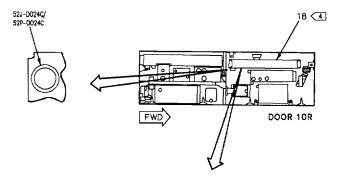


Figure 1. Countermeasures Set Locator (Sheet 5)

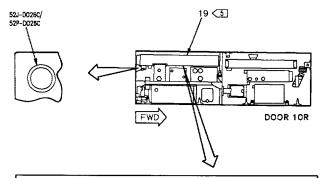
52A-C	159 NO. 8	CIRCUIT BREAKER/RELAY PANE	L ASSEMBLY
ZONE	REF DES	NOMENCLATURE	BUS
B1	76CBC007	INTER COMM	ESS 24/28VDC



NO. 7 CIRCUIT BREAKER/RELAY PANEL ASSEMBLY						
ZONE	REF DES	NOMENCLATURE	B∪S			
A11	64090011	ALQ-126	L115VACØA			
B11	64C8C012	ALQ-126	L115VACØB			
C11	64CBC013	ALQ-126	L115VACØC			



52A-D024		NO. 2 CIRCUIT BREAKER PANEL ASSEMBLY		
ZONE	REF DES	NOMENCLATURE	BUS	
D5	8080005	INT LTS	R28VDC	



 52A-D026
 NO. 4 CIRCUIT BREAKER PANEL ASSEMBLY

 ZONE
 REF DES
 NOMENCLATURE
 BUS

 C1D
 8C80005
 INT LTS
 R28/VOC

Figure 1. Countermeasures Set Locator (Sheet 7)

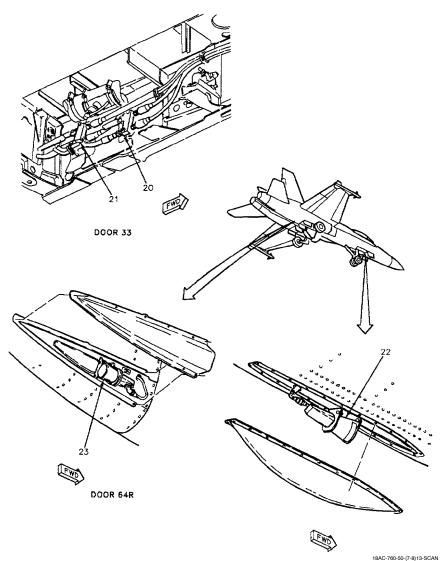


Figure 1. Countermeasures Set Locator (Sheet 8)

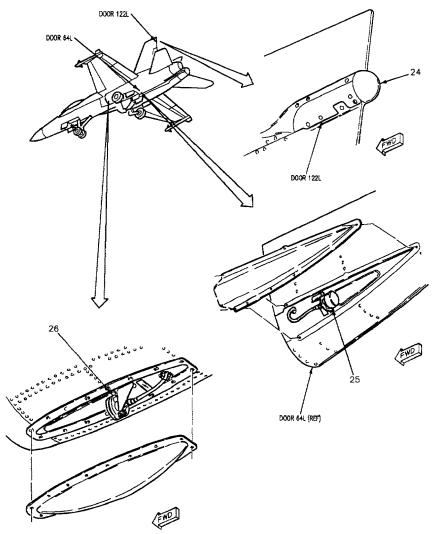


Figure 1. Countermeasures Set Locator (Sheet 9)

18AC-760-50-(7-9)13-SCAN

LEGEND

1.	AIRCRAFT CONNECTOR LOCATIONS ARE SHOWN IN A1-F18A()-WDM-000.
2	F/A-18B.
3	F/A-18A.
4	161353 THRU 161359.
5	161360 AND UP.
6	161702 AND UP.
7	F/A-18B 161354 THRU 161360.
8	F/A-18B 161704 AND UP.
9	161353 THRU 161528; ALSO 161702 THRU 163175 BEFORE F/A-18 AFC 50.

	NOMENCLATURE	INDEX NO.	REF DES
	AFT HIGH BAND ANTENNA AS-3421/ALQ-126	24	64E-S018
	AFT LOW BAND ANTENNA AS-3418/ALQ-126	25	64E-S006
	AFT MID BAND ANTENNA AS-3419/ALQ-126	23	64E-T008
	ARMAMENT COMPUTER CP-1342/AYQ-9(V)	12	61A-F001
6	COUNTERMEASURES COMPUTER CP-1293()/ALR-67(V)	10	62A-E006
	ECM CONTROL PANEL ASSEMBLY	4	52A-H087
	ECM COOLING AIR CONTROL VALVE	9	22L-E098
	FORWARD HIGH BAND ANTENNA AS-3385/ALQ-126	15	64E-B017
	FORWARD LOW BAND ANTENNA AS-3418/ALQ-126	26	64E-P005
	FORWARD MID BAND ANTENNA AS-3420/ALQ-126	22	64E-R007
	GND PWR CONTROL PANEL ASSEMBLY	3	1A-H004
	HIGH BAND COUPLER CU-2243/A	14	64DCF004
2	HIGH BAND SUPPRESSION FILTER F-1471/ALQ-126	7	64FLE010
3	HIGH BAND SUPPRESSION FILTER F-1471/ALQ-126	11	64FLE010
	INTERCOMMUNICATION AMPLIFIER-CONTROL	5	76A-H009
	INTERFERENCE BLANKER MX-9965/A	13	66A-F001
	INTR LT CONTROL BOX PANEL ASSEMBLY	2	8A-J002
	LH ADVISORY AND THREAT WARNING INDICATOR PANEL	1	52A-H073
	LOW BAND COUPLER CU-2264/A	20	64DCE003
	LOW BAND COUPLER CU-2264/A	21	64DCE002
4	NO. 2 CIRCUIT BREAKER PANEL ASSEMBLY	18	52A-D024
5	NO. 4 CIRCUIT BREAKER PANEL ASSEMBLY	19	52A-D026
	NO. 7 CIRCUIT BREAKER/RELAY PANEL ASSEMBLY	17	52A-C057

Figure 1. Countermeasures Set Locator (Sheet 10)

NOMENCLATURE	INDEX NO.	REF DES
NO. 8 CIRCUIT BREAKER/RELAY PANEL ASSEMBLY	16	52A-C159
RECEIVER-TRANSMITTER RT-1079B/ALQ-126	8	64A-E001
VOLUME CONTROL PANEL ASSEMBLY	6	76A-K032

Figure 1. Countermeasures Set Locator (Sheet 11)

SYSTEM SCHEMATICS

COUNTERMEASURES SET

Title	WP Number
Schematic - Receiver-Transmitter RT-1079A/ALQ-126	008 01 008 02

1 September 1992

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

SCHEMATIC - FUNCTIONAL

COUNTERMEASURES SET (ALQ-126A)

Reference Material

None

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Record of Applicable Technical Directives

Type/ Number	Date	Title and ECP No.	Date Incorp.	Remarks
F/A-18 AFC 50	16 Oct 84	Tactical Electronic Warfare Systems, ALR-67 Countermeasures, Modification of (ECP MDA-F/A-18-003R1 C1/C2/C3)	1 Oct 84	

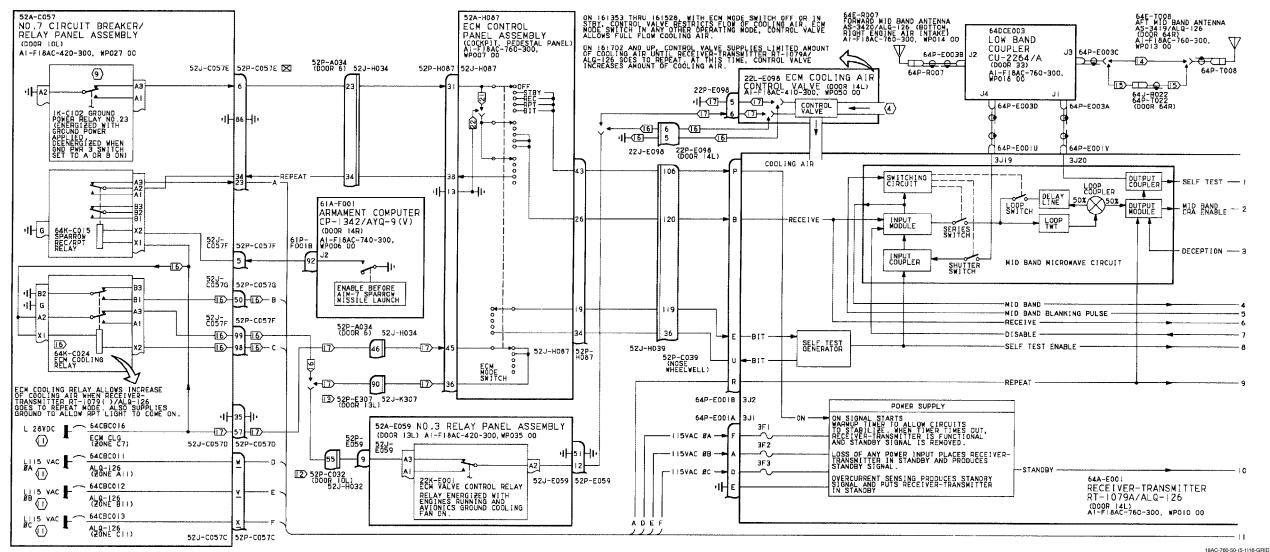
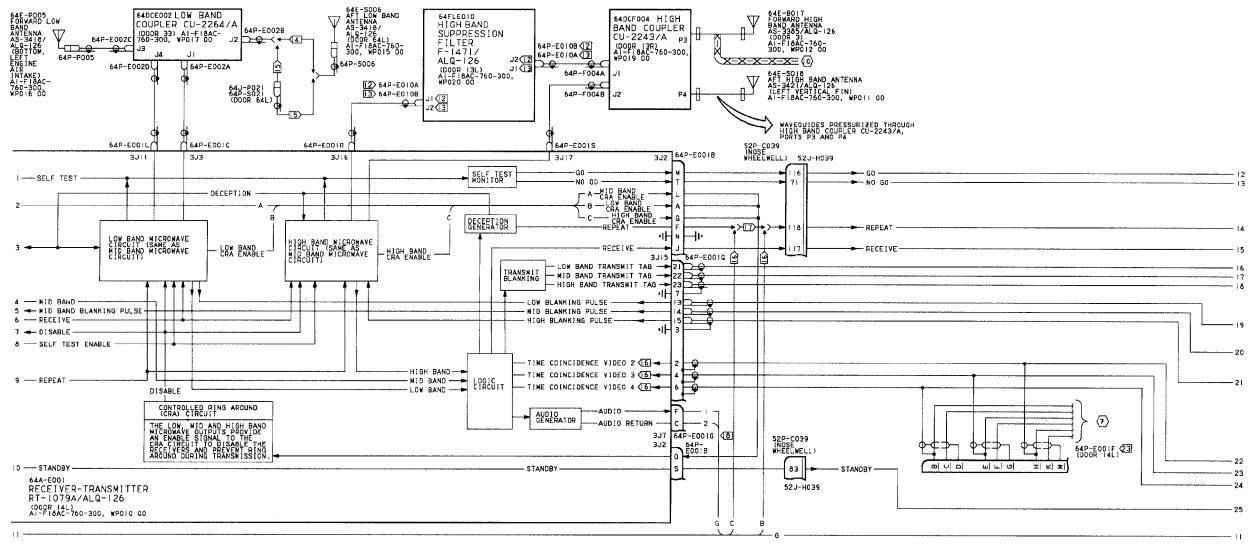


Figure 1. Receiver-Transmitter RT-1079A/ALQ-126 Functional Schematic (Sheet 1)



MID BLANKING PULSE

HIGH BLANKING PULSE

- TIME COINCIDENCE VIDEO 2

- TIME COINCIDENCE VIDEO 3

TIME COINCIDENCE VIDEO 4

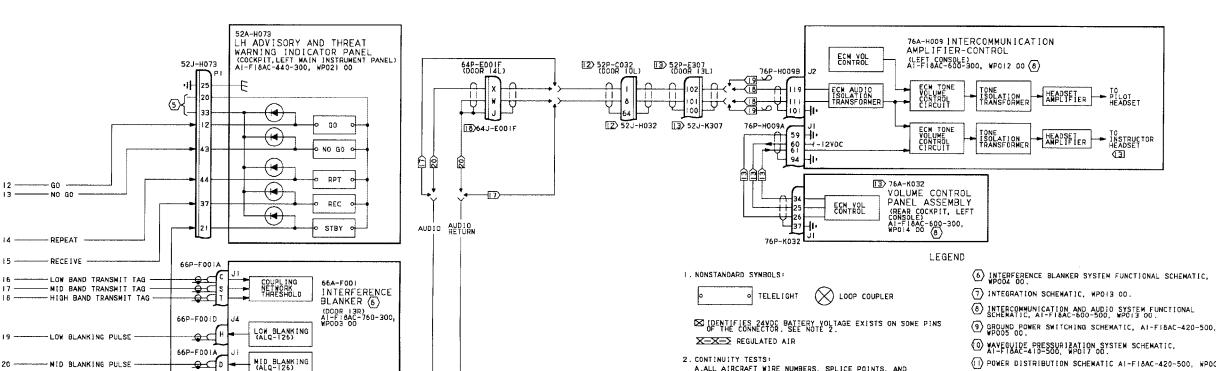
--- STANDBY

62P-E0068

62A-E006 COUNTERMEASURES (6) COMPUTER CP-+293()/ALR-67(V)

(DOOR |3L)A!-F!8AC-760-300, WP055 00

PROCESSOR (7)



- D. WHEN TESTING CONTINUITY, TEST FOR
- E.WHEN ELECTRICAL POWER IS OFF, 24VDC BATTERY VOLTAGE
 EXISTS ON SOME PINS ON CONNECTORS (IDENTIFIED BY 🔯)
 MAKE SURE MULTIMETER LEADS/JUMPER WIRES ARE INSTALLED
 ON CORRECT PINS WHEN TESTING FOR CONTINUITY
- 3. LINE UNDER LETTER (S) INDICATES LOWER CASE PIN LETTER.
- AVIONICS COOLING SYSTEM SCHEMATIC.
- (5) COCKPIT ADVISORY LIGHTS SCHEMATIC AT-FTBAC-440-500, WP006 00.

- (I) POWER DISTRIBUTION SCHEMATIC AI-FIBAC-420-500, WP005 00
- 12 F/A-18A.
- 3 F/A-18B
- [4] F/A-18A 161353 THRU 161705; ALSO F/A-18B 161354 THRU 161707.
- 15 F/A-18A 161706 AND UP: ALSO F/A-18B 161711 AND UP.
- 6 16 1702 AND UP
- 17 161353 THRU 161528
- [8] 161353 THRU 161528; ALSO 161702 THRU 163175 BEFORE F/A-18A AFC 50.
- 19 161702 THRU 163175 AFTER F/A-18A AFC 50.
- 20 161702 THRU 163175 BEFORE F/A-18A AFC 50.
- WIRE EXISTS ONLY WHEN ECH CONTROL PANEL ASSEMBLY PART NUMBER 74A800828-1001 OR 74A800828-1003 INSTALLED.
- WIRE EXISTS ONLY WHEN ECM CONTROL PANEL ASSEMBLY PART NUMBER 740800828-1001 INSTALLED.
- 23 PINS SHOWN ON THIS CONNECTOR ARE USED FOR RECEIVER-TRANSMITTER RT-1079B/ALQ-126 INSTALLATION ONLY.

18AC-760-50-(5-3)16-GRID

SYSTEM SCHEMATICS

FUNCTIONAL SCHEMATIC

COUNTERMEASURES SET (ALQ-126B)

Reference Material

None

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Record of Applicable Technical Directives

None

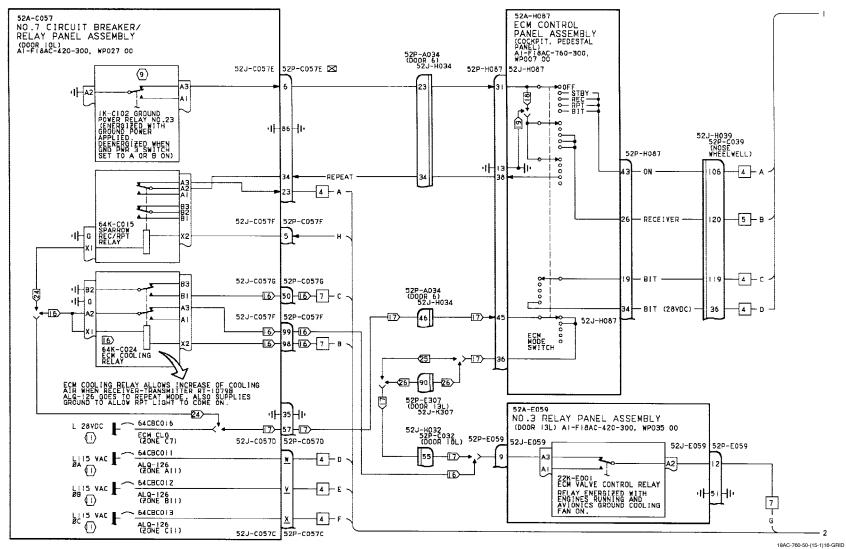
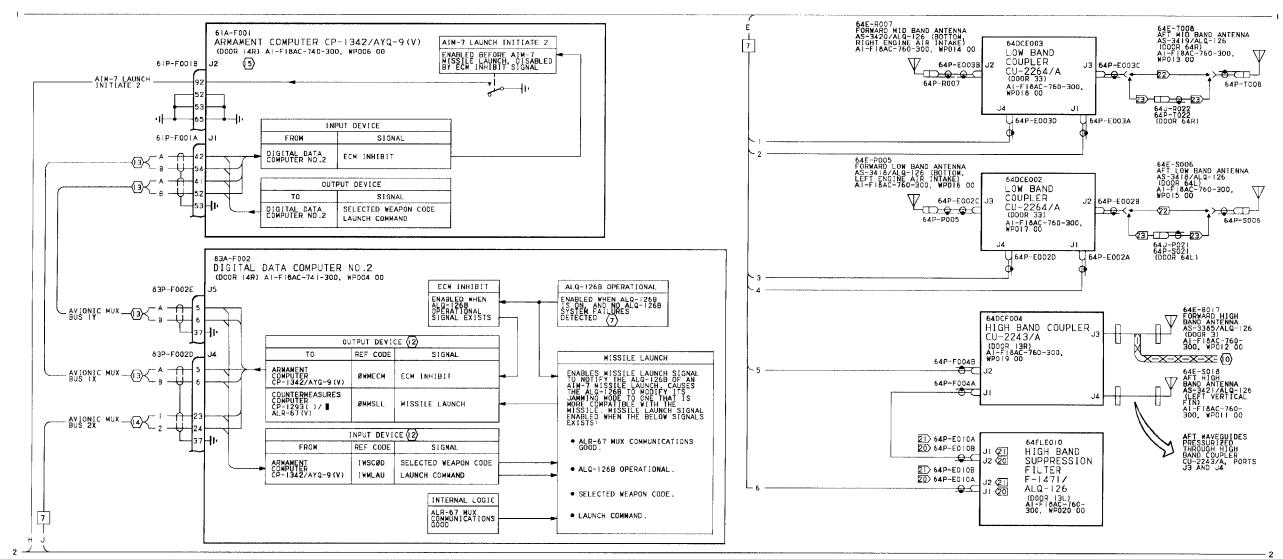
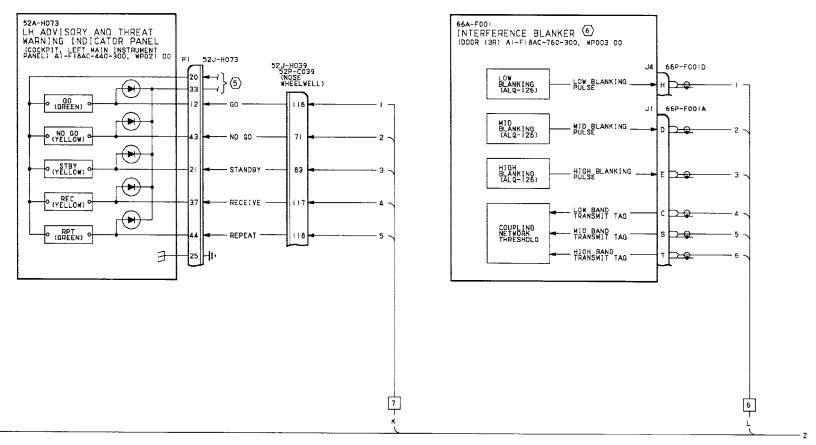
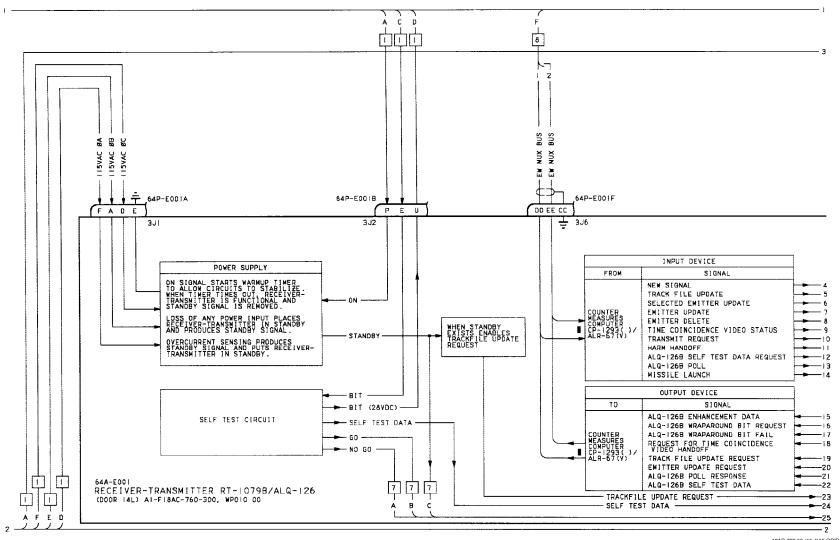


Figure 1.

Figure 1. Receiver-Transmitter RT-1079B/ALQ-126 Functional Schematic (Sheet 1)

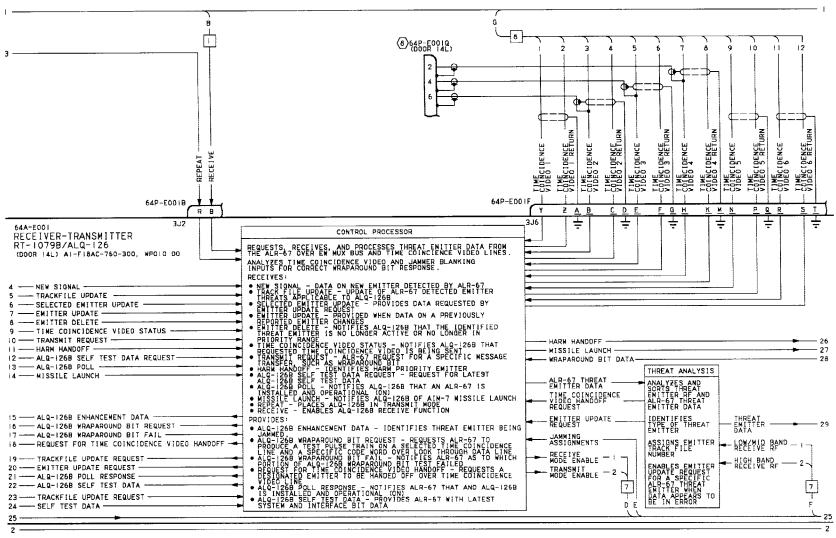




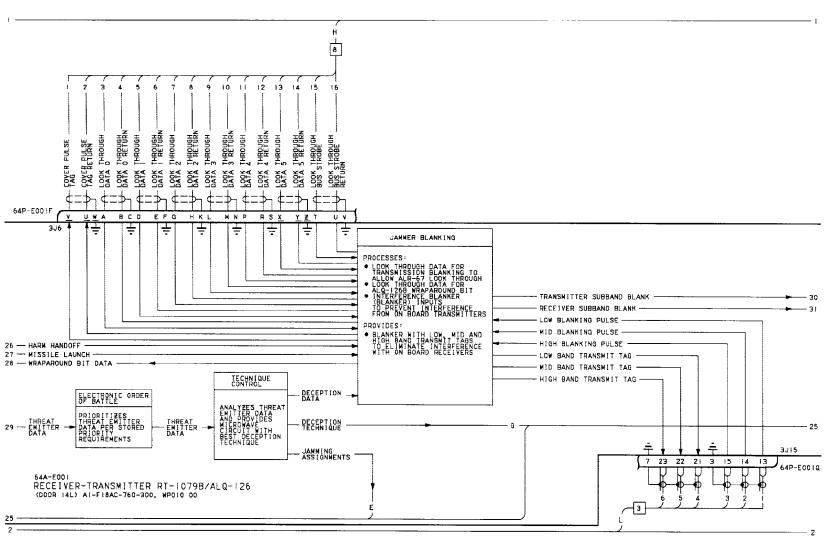


18AC-760-50-(15-4)16-GRID

Figure 1. Receiver-Transmitter RT-1079B/ALQ-126 Function Schematic (Sheet 4)

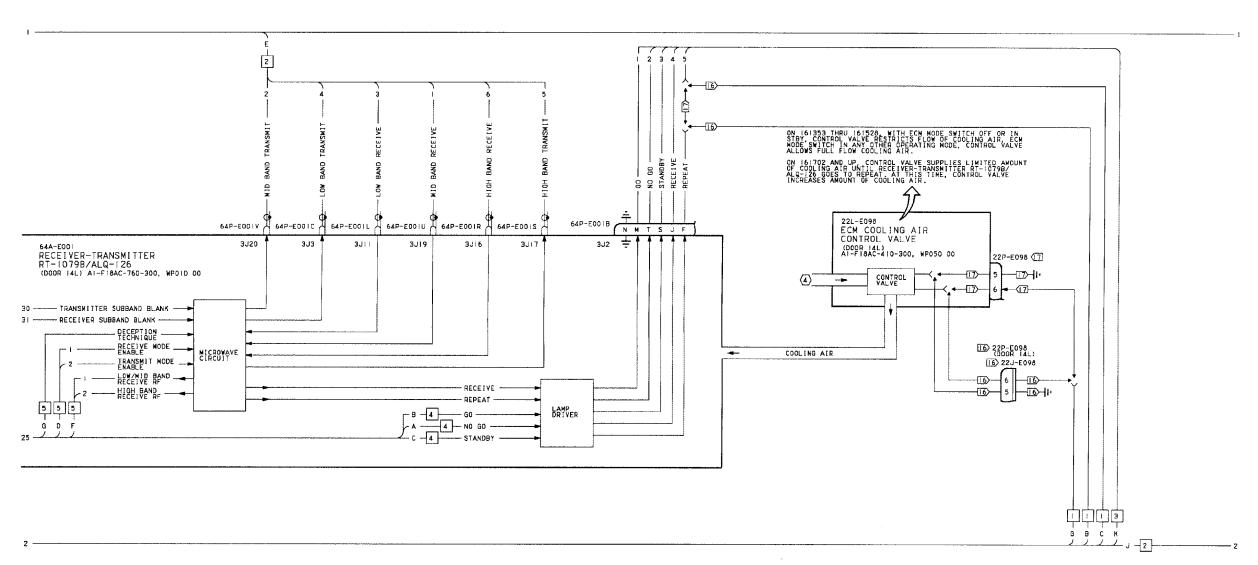


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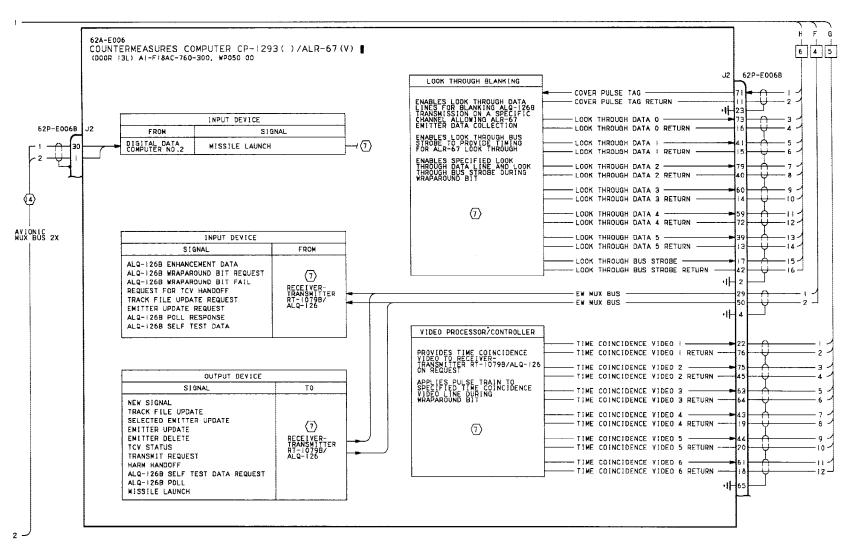


18AC-760-50-(15-6)16-GRID

Figure 1. Receiver-Transmitter RT-1079B/ALQ-126 Functional Schematic (Sheet 6)



A1-F18AC-760-500

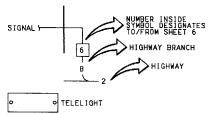


18AC-760-50-(15-8)16-GRID

008 02 Page 9 A1-F18AC-760-500 008 02 Page 10

LEGEND

I. NONSTANDARD SYMBOLS



X-X-X- REGULATED AIR

☑ IDENTIFIES 24YDC BATTERY VOLTAGE EXISTS ON SOME PINS OF THE CONNECTOR. SEE NOTE 2.

- 2. CONTINUITY TESTS:

 - A. ALL AIRCRAFT WIRE NUMBERS SPLICE POINTS, AND GROUND POINTS ARE SHOWN IN AI-FIBA() WDM-000.

 B. WHEN A LOW LEYEL CURRENT SWITCHING RELAY (IDENTIFIED BY ⊕) IS REMOVED FOR TROUBLESHOOTING. IDENTIFY RELAY AND SOCKET FOR CORRECT REINSTALLATION. DO NOT REPLACE LOW LEVEL CURRENT SWITCHING RELAY WITH ANY OTHER USED RELAY. IF RELAY IS DEFECTIVE, REPLACE WITH NEW RELAY.
 - C. DO NOT TEST LOW LEYEL DEVICES (SWITCHES/RELAY CONTACTS) FOR CONTINUITY WITH MULTIWETER ON THE RX! SCALE PIN TO PIN TESTS THAT DO NOT GO THROUGH SWITCHES/RELAY CONTACTS MAY USE THE RX! SCALE.
 - D. WHEN TESTING CONTINUITY, TEST FOR
 - SHORTS TO GROUND. SHORTS BETWEEN SURROUNDING PINS ON CONNECTORS. SHORTS BETWEEN SHIELD AND CONDUCTORS. SHIELD CONTINUITY.
 - E. WHEN ELECTRICAL POWER IS OFF, 24VDC BATTERY VOLTAGE EXISTS ON SOME PINS ON CONNECTORS (IDENTIFIED BY 🖾). MAKE SURE MULTI-METER LEADS/JUMPER WIRES ARE INSTALLED ON CORRECT PINS WHEN TESTING FOR CONTINUITY.
- 3. LINE UNDER LETTER (S) INDICATES LOWER CASE PIN LETTER.
- 4 AVIONICS COOLING SYSTEM SCHEMATIC-EXCEPT COCKPIT, AI-FIBAC-410-500,
- (5) COCKPIT ADVISORY LIGHTS SCHEMATIC, A1-F18AC-440-500, WP006 DD.
- INTERFERENCE BLANKER SYSTEM FUNCTIONAL SCHEMATIC, WP004 00.
- (7) INTEGRATION SCHEMATIC, WPO13 00.
- ⟨₹) USED WHEN RECEIVER-TRANSMITTER RT-1079A/ALQ-126 IS INSTALLED.
- (9) GROUND POWER SWITCHING SCHEMATIC, AI-FIBAC-420-500, WP005 00.

- (1) WAYEGUIDE PRESSURIZATION SYSTEM SCHEMATIC, AI-FIBAC-410-500,
- (II) POWER DISTRIBUTION SCHEMATIC, AI-FIBAC-420-500, WP005 00
- FOR LOGICY PLASMANS RELATING TO REF CODE THEFER TO AL-FLOAD TO ALL FLOAD TO ALL FLO
- (13) AVIONIC MUX CHANNEL I SCHEMATIC, AI-FI8AC-741-500, WP004 00.
- (4) AVIONIC MUX CHANNEL 2 SCHEMATIC, AI-FIBAC-741-500, WP005 00
- AIM-7 SPARROW AVIONIC INTERFACE SCHEMATIC, AI-FIBAC-740-510,
- [6] 161702 AND UP.
- [7] F/A-18A 161353 THRU 161528; ALSO F/A-18B 161354 THRU 161360.
- HEN ECM CONTROL PANEL ASSEMBLY PART NUMBER 74A800828-1001, 74A800828-1003 OR 74A800828-1005 INSTALLED.
- 19 WIRE EXISTS ONLY WHEN ECM CONTROL PANEL ASSEMBLY PART NUMBER 746800828-1001 INSTALLED.
- 20> F/A-18A.
- 21) F/A-18B
- 22 161353 THRU 161705; ALSO F/A-18B 161707.
- 23 F/A-18A 161706 AND UP; ALSO F/A-18B 161711 AND UP.
- 24) 161353 AND UP.
- 25 F/A-18A 161353 THRU 161528
- 26) F/A-18B 161354 THRU 161360

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

LOCATOR

COUNTERMEASURES WARNING AND CONTROL SYSTEM

EFFECTIVITY: 161702 AND UP

Reference Material

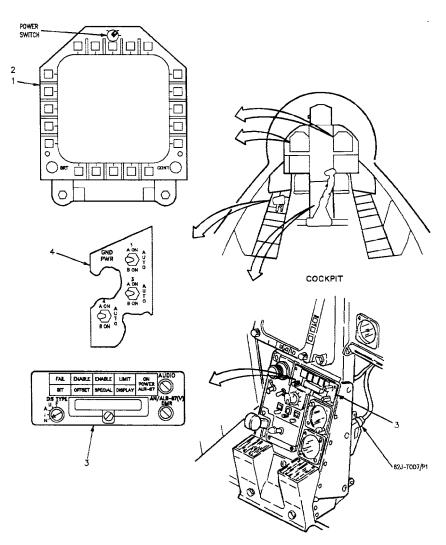
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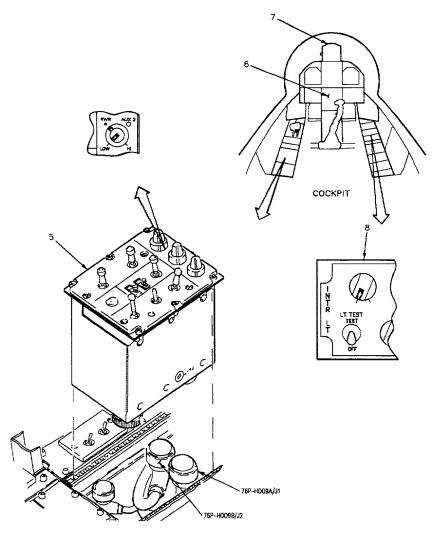
Record of Applicable Technical Directives

None



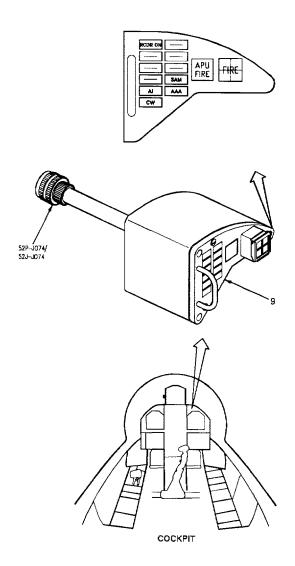
18AC-760-50-(9-1)14-SCAN

Figure 1. Countermeasures Warning and Control System Locator (Sheet 1)



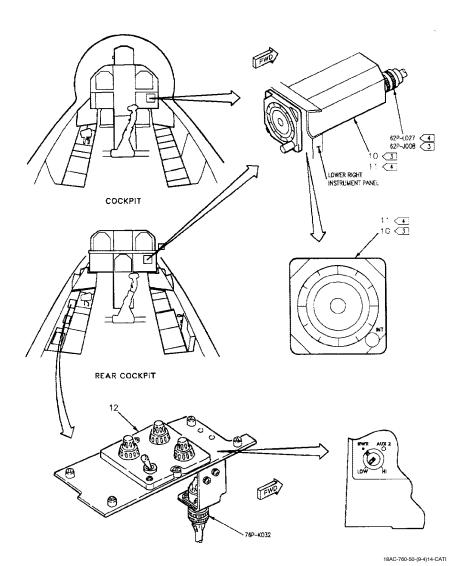
18AC-760-50-(9-2)14-SCAN

Figure 1. Countermeasures Warning and Control System Locator (Sheet 2)



18AC-760-50-(9-3)14-CATI

Figure 1. Countermeasures Warning and Control System Locator (Sheet 3)



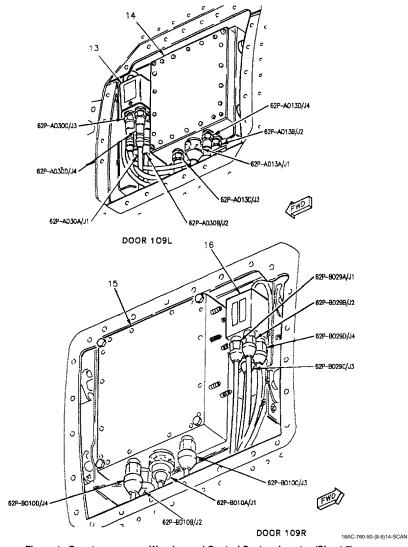


Figure 1. Countermeasures Warning and Control System Locator (Sheet 5)

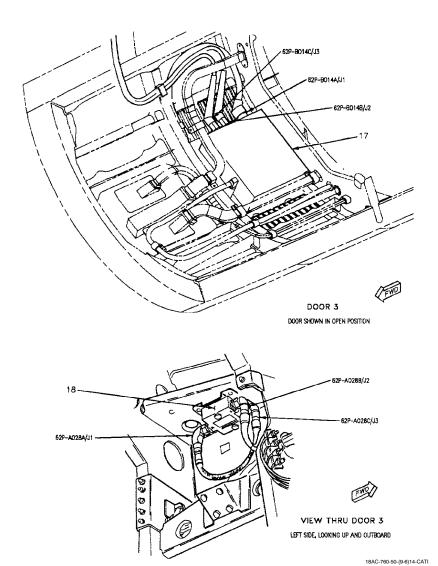
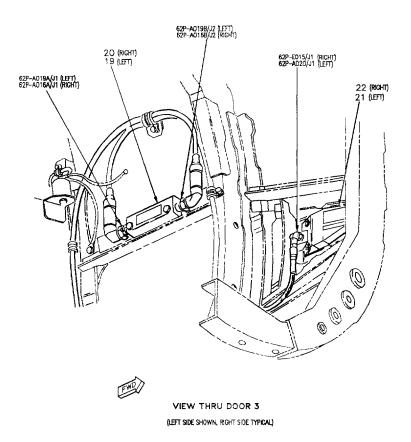
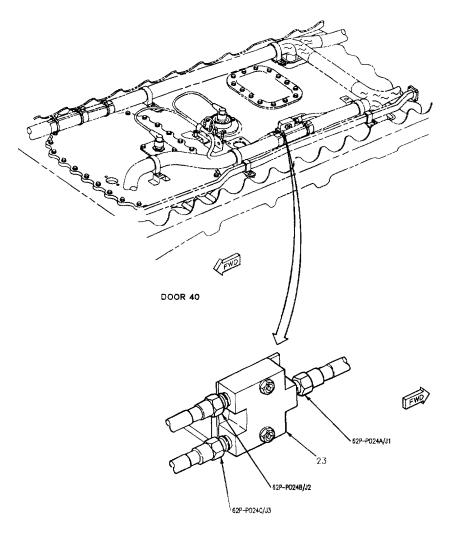


Figure 1. Countermeasures Warning and Control System Locator (Sheet 6)



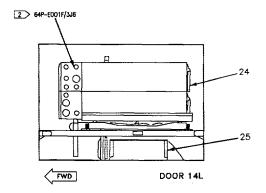
18AC-760-50-(9-7)14-SCAN

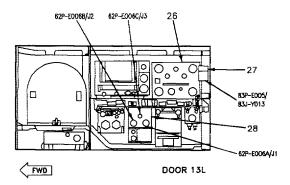
Figure 1. Countermeasures Warning and Control System Locator (Sheet 7)



18AC-760-50-(9-8)14-CATI

Figure 1. Countermeasures Warning and Control System Locator (Sheet 8)





18AC-760-50-(9-9)14-CATI

Figure 1. Countermeasures Warning and Control System Locator (Sheet 9)

62P-T035C/J3 (RIGHT) 62P-S036C/J3 (LETT)

32 (RIGHT) 31 (LEFT)

SPLICE AREA

37 (LEFT) -38 (RIGHT)

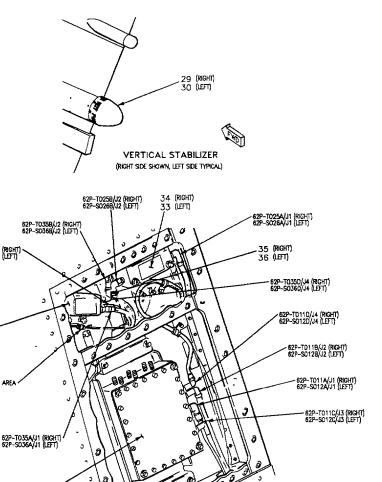
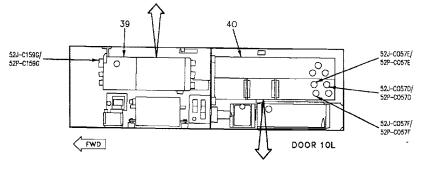


Figure 1. Countermeasures Warning and Control System Locator (Sheet 10)

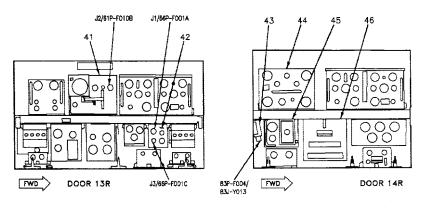
DOOR 124R, DOOR 124L (RIGHT SIDE SHOWN, LEFT SIDE TYPICAL)

18AC-760-50-(9-10)14-SCAN

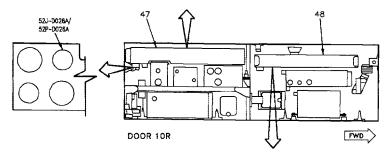
52A-C	159	NO. 8 CIRCUIT BREAKER/RELAY PANEL ASSEMBLY	
ZONE	REF DES	NOMENCLATURE	BUS
B1	76CBC027	INTER COMM	ESS 24/28VDC
D2	85CBC004	MSDRS	MAINT 24/28VD
D12	80090006	MMD	L 115VAC ØC
E12	80CBC005	MMD	L 115VAC ØB
F12	80090004	MMD	L 115VAC ØA



ZONE	REF DES	NOMENCLATURE	BUS
A11	64CBC011	ALQ-126/165	L 115VAC ØA
A20	83CBC006	MISSION COMP NO. 1	L 115VAC ØA
A27	62CBC001	ALR-67	L 115VAC ØA
A28	62CBC002	ALR-67	L 115VAC ØA
B9	62CBC005	ALR-67 IND CONT/FLTR	L 28VDC
B11	64CBC012	ALQ-126/165	L 115VAC ØB
B20	83CBC007	MISSION COMP NO. 1	L 115VAC ØB
B26	62080003	ALR-67	L 115VAC ØB
C11	64CBC013	ALQ-126/165	L 115VAC ØC
C20	83080008	MISSION COMP NO. 1	L 115VAC ØC
C26	62CBC004	ALR-67	L 115VAC ØC



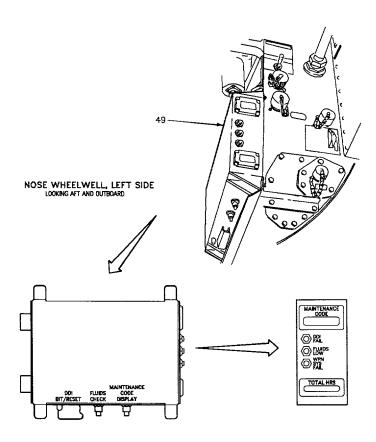
52A-D026 NO. 4 CIRCUIT BREAKER PANEL ASSEMBLY		
REF DES	NOMENCLATURE	BUS
82CBD005	csc	R 28VDC
		R 28VDC R 28VDC
	REF DES	RET DES NOMENCIATURE 82090005 CSC 76090025 INTERCOM



ZONE	REF DES	NOMENCLATURE	BUS
A11	820BD002	CSC	R 115VAC ØA
A12	66CBD002	BLANKER	R 115VAC ØA
A17	80CBD007	MFD	R 115VAC ØA
B11	82CBD003	CSC	R 115VAC ØB
B17	BOCEDOCE	MFD	R 115VAC ØB
C11	82080004	CSC	R 115VAC ØC
D7	800BD009	MFD	R 115VAC Ø

18AC-760-50-(9-12)14-CATI

Figure 1. Countermeasures Warning and Control System Locator (Sheet 12)



18AC-760-50-(9-13)14-CATI

Figure 1. Countermeasures Warning and Control System Locator (Sheet 13)

NOMENCLATURE	INDEX NO.	REF DES	
ADAPTER	35	62CPT035B	1
ADAPTER	36	62CPS036B	ı
ARMAMENT COMPUTER CP-1342/AYQ-9(V)	45	61A-F001	ı
COMMAND LAUNCH COMPUTER CP-1001()/AWG	47	61A-F010	ı
CONTROL-INDICATOR C-10250/ALR-67(V)	2	62A-J007	ı
COUNTERMEASURES COMPUTER CP-1293()/ALR/67(V)	28	62A-E006	
DIGITAL DATA COMPUTER NO. 1	26	83A-E001	ı
DIGITAL DATA COMPUTER NO. 2	44	83A-F002	ı
DIGITAL DISPLAY INDICATOR ID-2150/ASM-612	49	85A-G003	ı
FORWARD AZIMUTH INDICATOR IP-1276/ALR-67(V)	10	62A-J008	ı
FORWARD RADAR RECEIVER COUPLER CU-2292/ ALR-67(V)	17	62A-A028	
GND PWR CONTROL PANEL ASSEMBLY	1	1A-H004	
HEAD-UP DISPLAY UNIT AN/AVQ-28	5	79A-J001	ı
HORIZONTAL INDICATOR IP-1350/A	6	80A-J003	ı
INTEGRATED ANTENNA AS-3190A/ALR-67(V)	18	62A-B014	ı
INTERCOMMUNICATION AMPLIFIER-CONTROL	8	76A-H009	ı
INTERFERENCE BLANKER MX-9965/A	48	66A-F001	
INTR LT CONTROL BOX PANEL ASSEMBLY	7	8A-J002	
LEFT DIGITAL DISPLAY INDICATOR IP-1317()	3	80A-H001	
LEFT FORWARD ANTENNA-RADOME AS-3360/ALR	22	62E-A020	
LEFT FORWARD BAND PASS FILTER F-1539/ALR-67(V)	19	62FLA019	
LEFT FORWARD RADAR RECEIVER R-2148A/ALR-67(V)	14	62A-A013	
LEFT FORWARD RADIO FREQUENCY TRANSMISSION SWITCH SA-2362/ALR-67(V)	13	62S-A030	
LEFT MUX BUS IMPEDANCE MATCHING NETWORK	27	83A-Y013	
LEFT REAR ANTENNA-RADOME AS-3306/ALR	30	62E-S018	
LEFT REAR BAND PASS FILTER F-1539/ALR/67(V)	33	62FLS026	
LEFT REAR RADAR RECEIVER R-2148A/ALR-67(V)	37	62A-S012	I
LEFT REAR RADIO FREQUENCY TRANSMISSION SWITCH SA-2362/ALR-67(V)	31	62S-S036	
NO. 2 CIRCUIT BREAKER PANEL ASSEMBLY	42	52A-D024	I
NO. 4 CIRCUIT BREAKER PANEL ASSEMBLY	41	52A-D026	
		•	

Figure 1. Countermeasures Warning and Control System Locator (Sheet 14)

	NOMENCLATURE	INDEX NO.	REF DES
	NO. 7 CIRCUIT BREAKER/RELAY PANEL ASSEMBLY	40	52A-C057
	NO. 8 CIRCUIT BREAKER PANEL ASSEMBLY	39	52A-C159
	RADAR RECEIVER R-2055A/ALR-67(V)	25	62A-E009
4	REAR AZIMUTH INDICATOR IP-1276/ALR-67(V)	11	62A-L027
	REAR RADAR RECEIVER COUPLER CU-2292/ALR-67(V)	23	62A-P024
	RECEIVER-TRANSMITTER RT-1079()/ALQ-126	24	64A-E001
	RH ADVISORY AND THREAT WARNING INDICATOR PANEL	9	52A-J074
	RIGHT DIGITAL DISPLAY INDICATOR IP-1317()	4	80A-J002
	RIGHT FORWARD ANTENNA-RADOME AS-3359/ALR	21	62E-B015
	RIGHT FORWARD BAND PASS FILTER F-1539/ALR-67(V)	20	62FLB016
	RIGHT FORWARD RADAR RECEIVER R-2148A/ALR-67(V)	15	62A-B010
	RIGHT FORWARD RADIO FREQUENCY TRANSMISSION SWITCH SA-2362/ALR-67(V)	16	62S-B029
	RIGHT MUX BUS IMPEDANCE MATCHING NETWORK	46	83A-Y013
	RIGHT REAR ANTENNA-RADOME AS-3306/ALR	29	62E-T017
	RIGHT REAR BAND PASS FILTER F-1529/ALR-67(V)	34	62FLT025
	RIGHT REAR RADAR RECEIVER R-2148A/ALR-67(V)	38	62A-T011
	RIGHT REAR RADIO FREQUENCY TRANSMISSION SWITCH SA-2362/ALR-67	32	62S-T035
	SIGNAL DATA RECORDER RO-508/ASM-612	43	85A-F001
	VOLUME CONTROL PANEL ASSEMBLY	12	76A-K032

LEGEND

1.	AIRCRAFT CONNECTOR LOCATIONS ARE SHOWN IN A1-F18A()-WDM-000.
2	USED ON RECEIVER-TRANSMITTER RT-1079B/ALQ-126 ONLY.
3	F/A-18A.
4	F/A-18B.

Figure 1. Countermeasures Warning and Control System Locator (Sheet 15)

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATIC

SCHEMATIC - INTERCONNECT

COUNTERMEASURES WARNING AND CONTROL SYSTEM

EFFECTIVITY: 161702 AND UP

This WP supersedes WP010 00, dated 1 December 1993

Reference Material

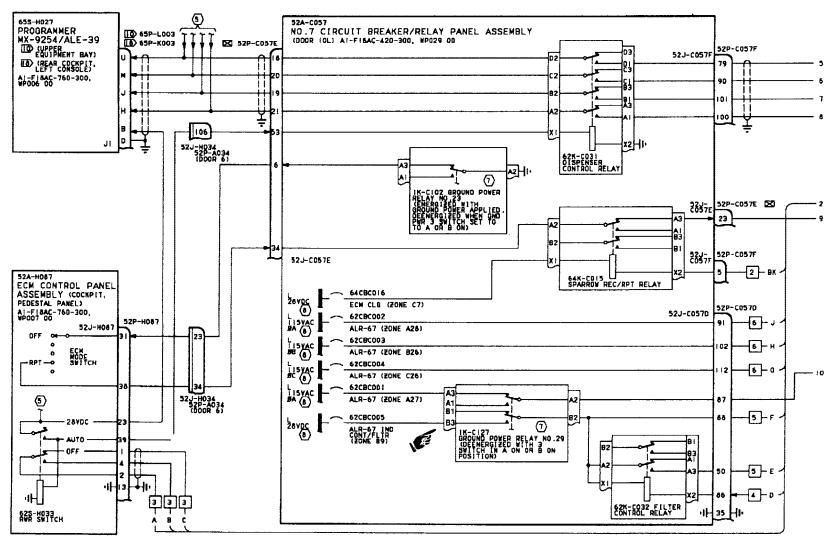
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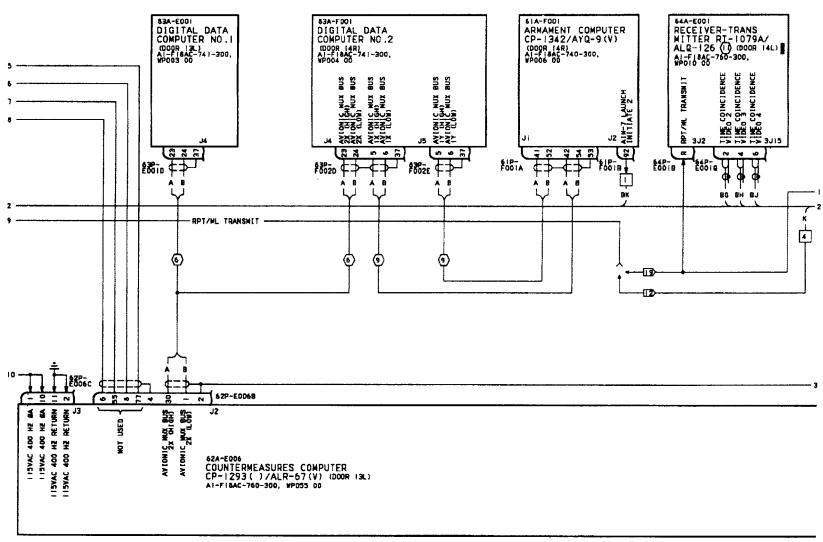
Record of Applicable Technical Directives

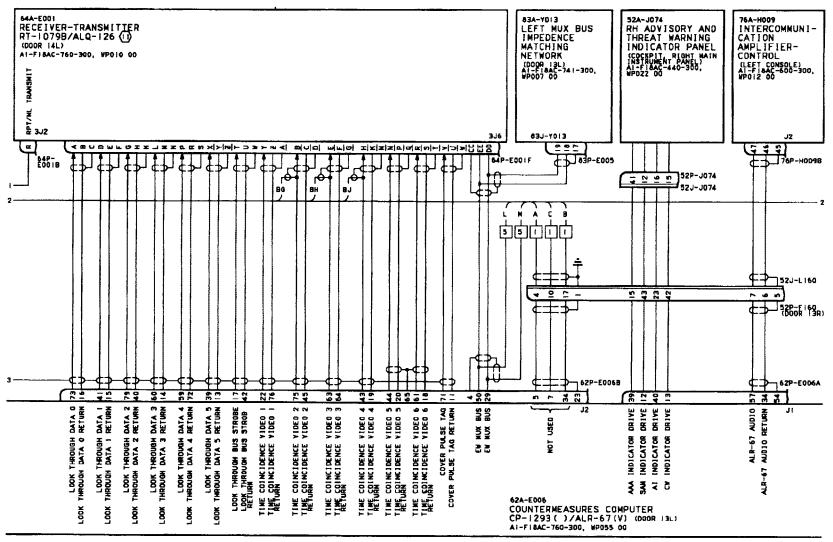
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F/A-18 AFC-158	Oct 91	Correction of AN/ALR-67 and AN/ ALR-126B Wiring (ECP RAMEC NO- RIS-22-90)	1 Sept 92	-

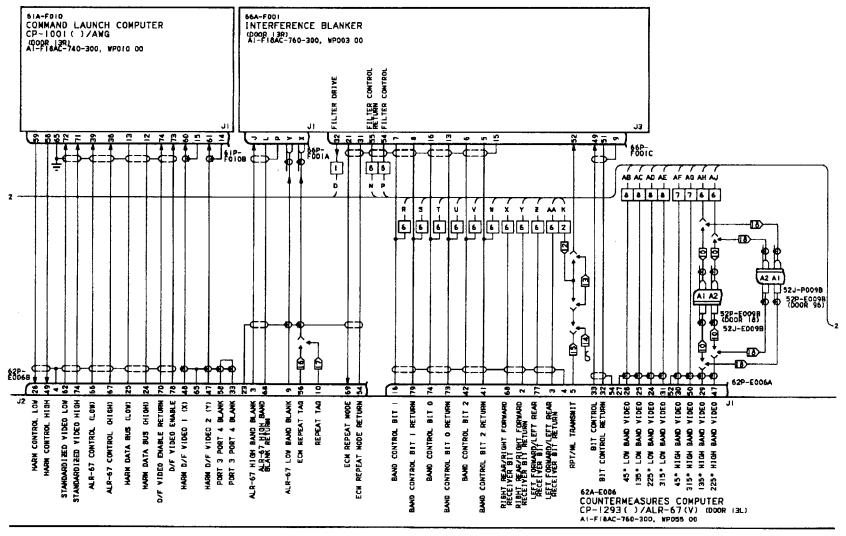


18AC-760-50-(8-1)16-GRID

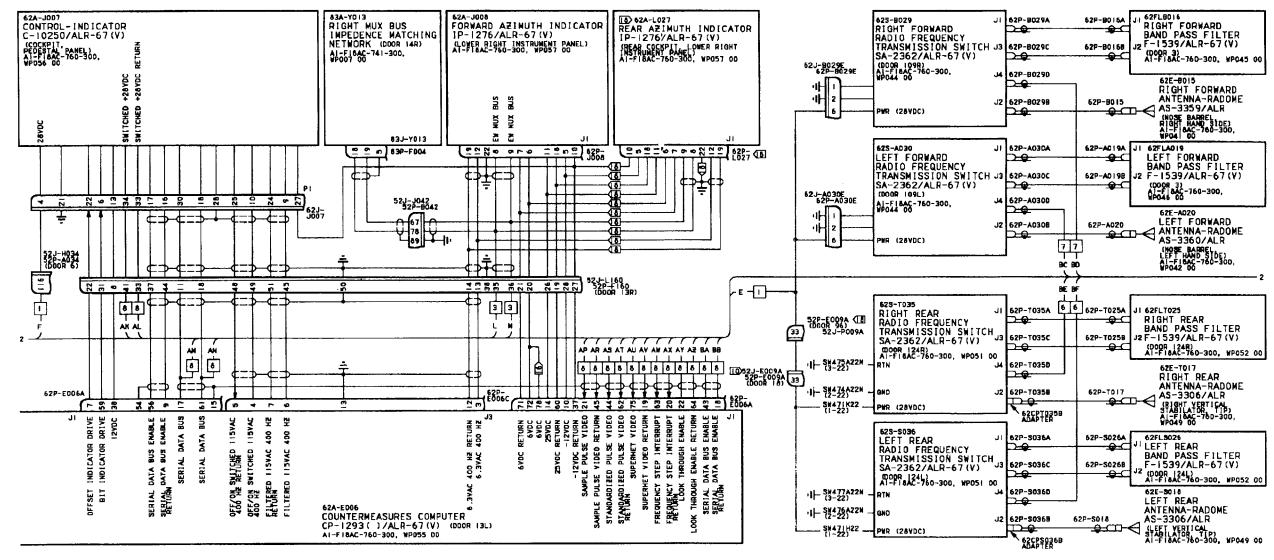
Figure 1. Countermeasures Warning and Control System Interconnect Schematic (Sheet 1)

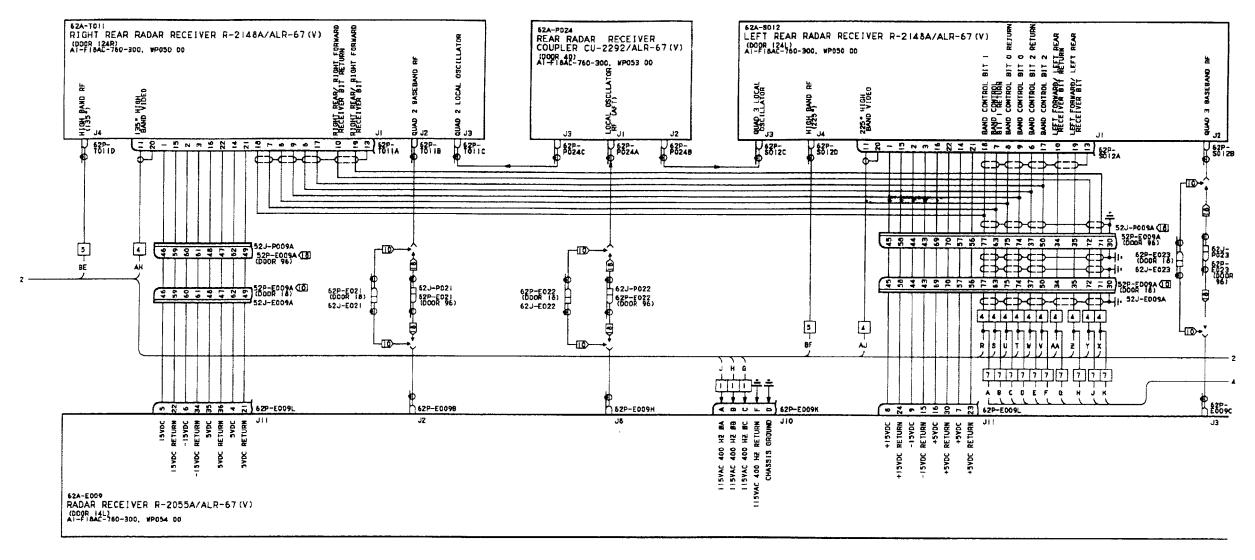


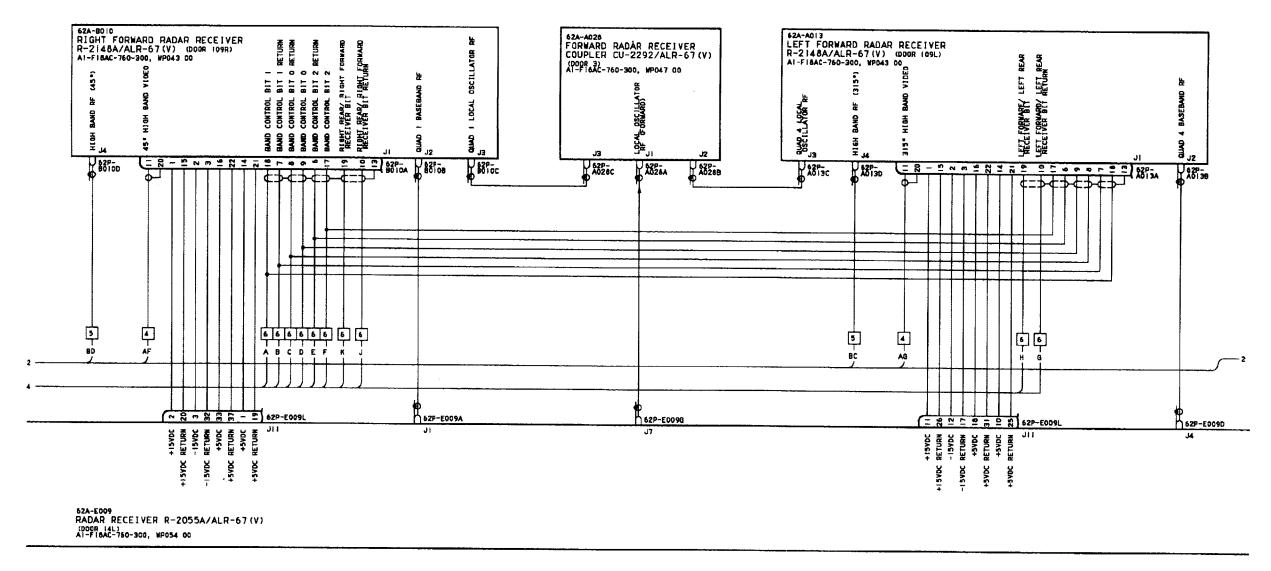


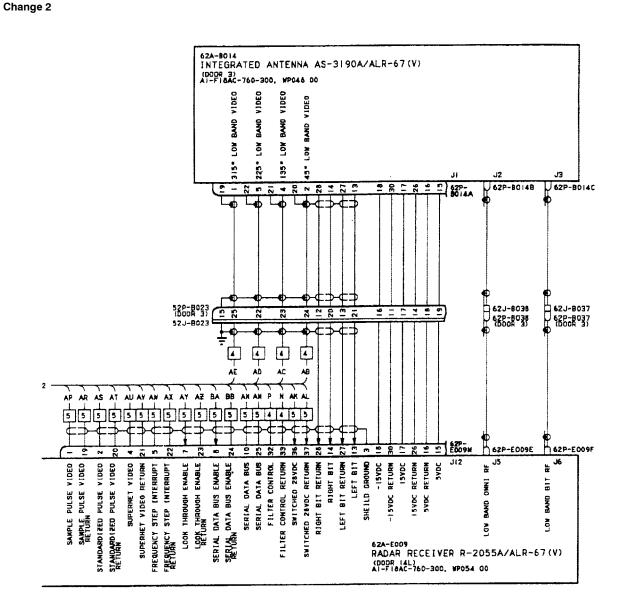


Change 2









LEGEND

- . HONSTANDARD SYMBOLS

 ① IDENTIFIES RELAY USED TO SWITCH LOW LEVEL CURRENT.
 SEE NOTE 2.
- DE IDENTIFIES 24VDC BATTERY VOLTAGE EXISTS ON SOME PINS OF THE CONNECTOR, SEE NOTE 2.



- 2. CONTINUITY TESTS
- A. ALL A) HCRAFT WIRE NUMBERS, SPLICE POINTS AND GROUND POINTS ARE SHOWN IN AI-FIGA() NDH-DOO.

- D. WHEN TESTING CONTINUITY, TEST FOR
 - (1) SHORTS TO BROUND
 - (2) SHORTS BETWEEN SURROUNDING PINS ON CONNECTORS.
 - (3) SHORTS BETWEEN SHIELD AND CONNECTORS
- . WHEN ELECTRICAL POWER IS OFF, ZAVDC BATTERY VOLTAGE EXISTS ON SOME PINS ON CONNECTORS (IDENTIFIED BY SOL) HARE SURE MULTIMETER ELECTRICATE VIRES ARE INSTALLED ON CORRECT PINS WHEN TESTING FOR CONTINUTY.
- 3. LINE UNDER LETTER (S) INDICATES LOWER CASE PIN LETTER
- 4. NONSTANDARD ABBREVIATIONS
- BIT BUILT-IN TEST
- HARN HIGHSPEED ANTI-RADIATION MISSLE

- (5) COUNTERMEASURES DISPENSING SYSTEM FUNCTIONAL SCHEMATIC, MP006 00.
- (6) AVIDNIC MUX CHANNEL 2 SCHEMATIC, AI-FIBAC-741-500,
- 7 BROUND POWER SWITCHING SCHEMATIC, AI-FIBAC-420-500,
- (8) POWER DISTRIBUTION SCHEMATIC AI-FIBAC-420-500,
- 9 AYIONIC MUX CHANNEL : SCHEMATIC, AI-FIBAC-741-500,
- ID F/A-18A
- (1) RECEIVER-TRANSMITTER RT-1079A/ALQ-126 AND RT-1079B/ ALQ-126 ANE ALTERNATE CONFIGURATIONS OF THE COUNTER-MEASURES SET.
- 12 161737 AND UP
- 13 161702 THRU 161736
- 162853 AND UP: ALSO 161702 THRU 163175 AFTER
- (5) 161702 THRU 163175 BEFORE F/A-16A AFC 50.
- 161737 AND UP: ALSO 161702 THRU 161736 AFTER F/A-18A AFC 156.
- 15:702 THRU 16:736 BEFORE F/A-18A AFC 158.
- DF/A-18B
- 15 161702 AND UP

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

SCHEMATIC - POWER INTERFACE

COUNTERMEASURES WARNING AND CONTROL SYSTEM

EFFECTIVITY: 161702 AND UP

This WP supersedes WP011 00, dated 1 September 1992.

Reference Material

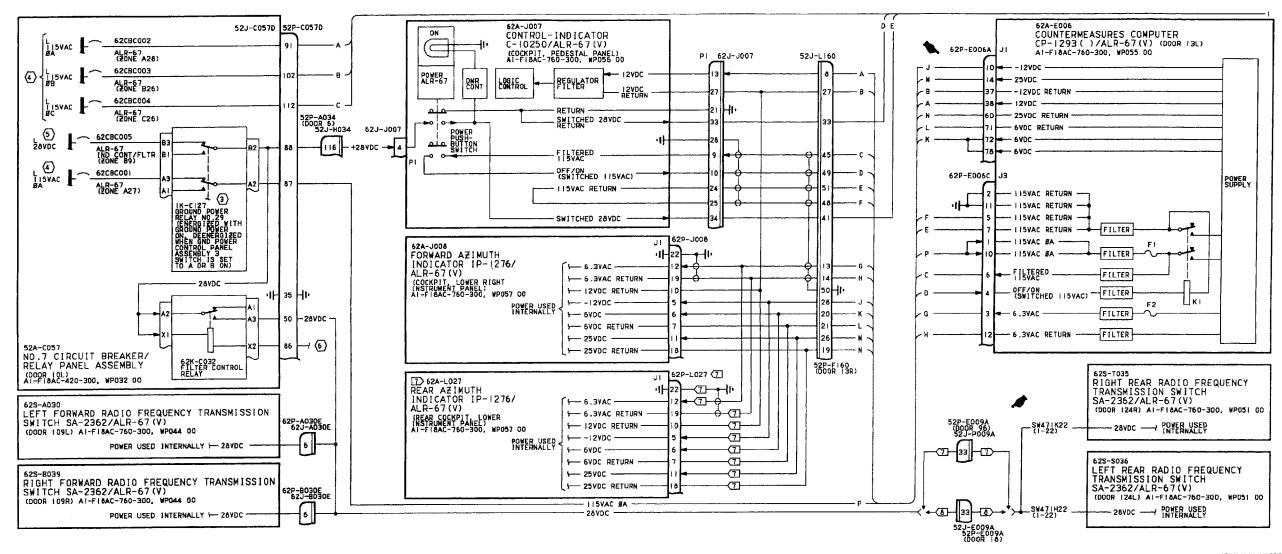
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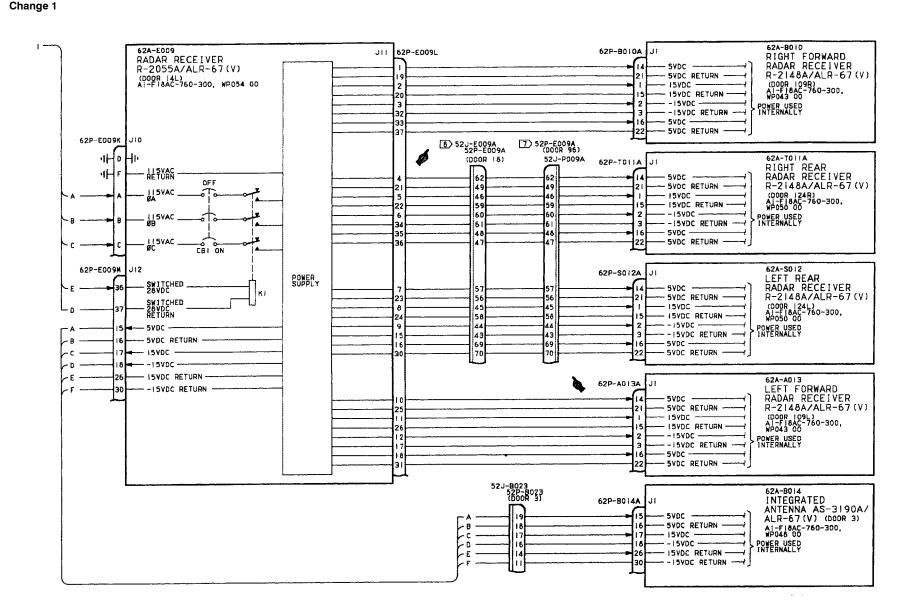
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Record of Applicable Technical Directives

None





LEGEND

- I . NONSTANDARD SYMBOL:
 - # IDENTIFIES RELAY USED TO SWITCH LOW LEVEL CURRENT . SEE NOTE 2 .
- 2. CONTINUITY TESTS:
 - . ALL AIRCRAFT WIRE NUMBERS, SPLICE POINTS AND
- B. WHEN A LOW LEVEL CURRENT SWITCHING RELAY (IDENTIFIED BY A) IS REMOVED FOR TROUBLESHOOTING. IDENTIFIED BY A COLOR OF THE CONTROL OF THE COLOR OF
- DO NOT IEST LOW LEVEL DEVICES (SWITCHES/RELAY CONTACTS)
 FOR CONTINUITY WITH MUST WEETER ON THE RXI SCALE. PIN TO
 PIN TESTS THAT DO NOT GO THROUGH SWITCHES/RELAY CONTACTS
 MAY USE THE RXI SCALE.
- D. WHEN TESTING CONTINUITY, TEST FOR:
- (1) SHORTS TO GROUND.
- (2) SHORTS BETWEEN SURROUNDING PINS ON CONNECTORS.
- (3) SHORTS BETWEEN SHIELD AND CONNECTORS.
- (4) SHIELD CONTINUITY.
- (3) GROUND POWER SWITCHING SCHEMATIC, AI-FIBAC-420-500, WP005 00.
- (4) AC POWER SCHEMATIC, AI-FIBAC-420-500, WPD03 00.
- 5 DC POWER SCHEMATIC, AI-FIBAC-420-500, WP004 00.
- (6) RF DETECTION AND CONVERSION SCHEMATIC, WPO12 00.
- 7 F/A-188
- B) F/A-18A.

18AC-760-50-(10-2)17-GRID

Figure 1. Countermeasures Warning and Control System Power Interface Schematic (Sheet 2)

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

SCHEMATIC - RF DETECTION AND CONVERSION

COUNTERMEASURES WARNING AND CONTROL SYSTEM

EFFECTIVITY: 161702 AND UP

This WP supersedes WP012 00, dated 1 September 1992.

Reference Material

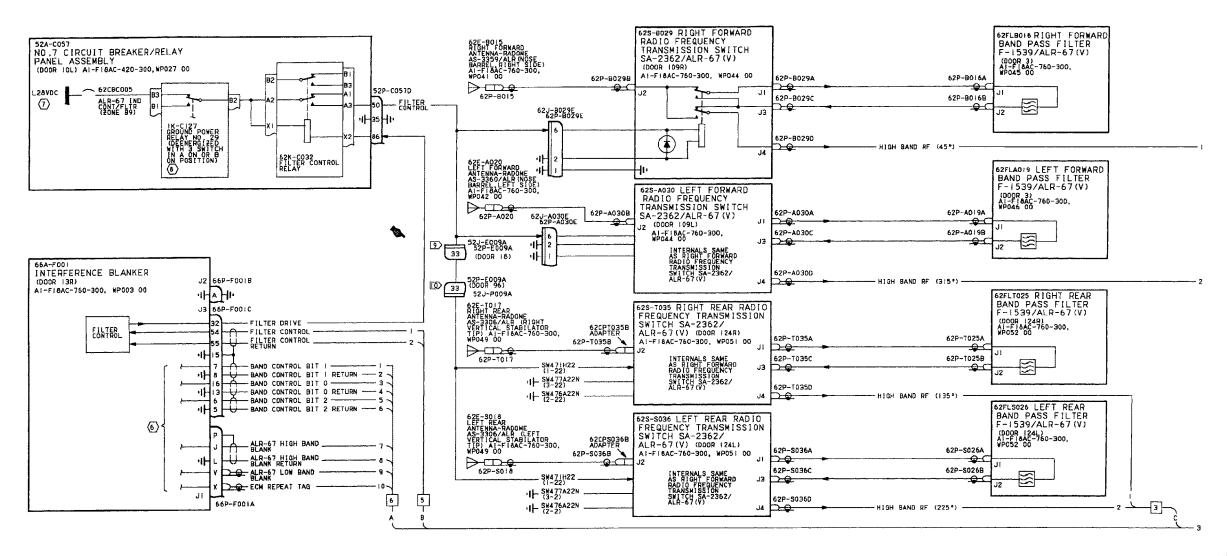
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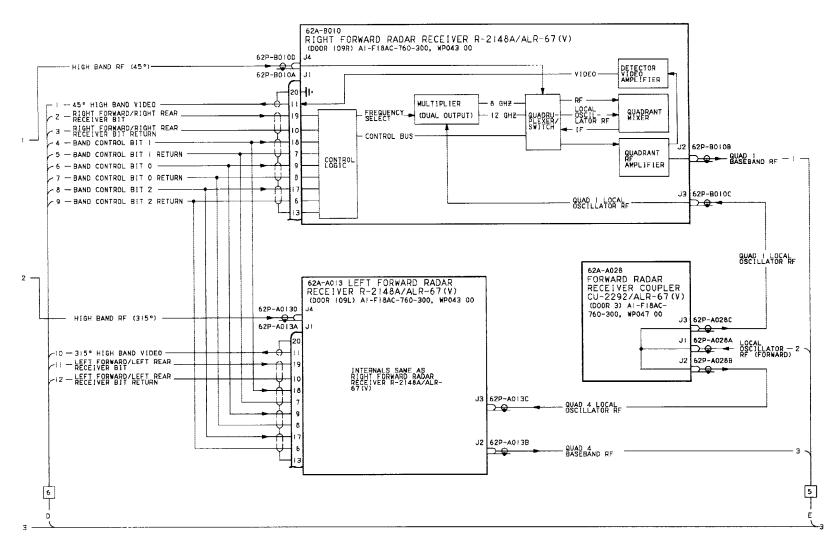
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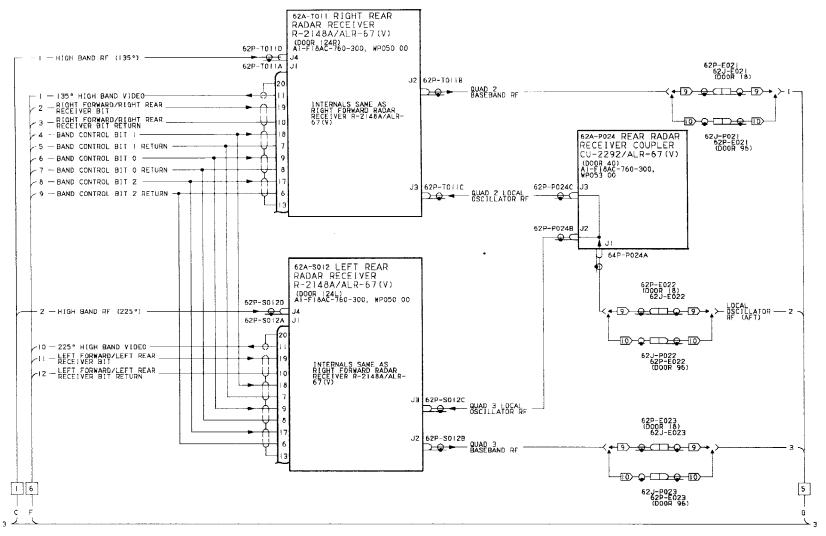
Record of Applicable Technical Directives

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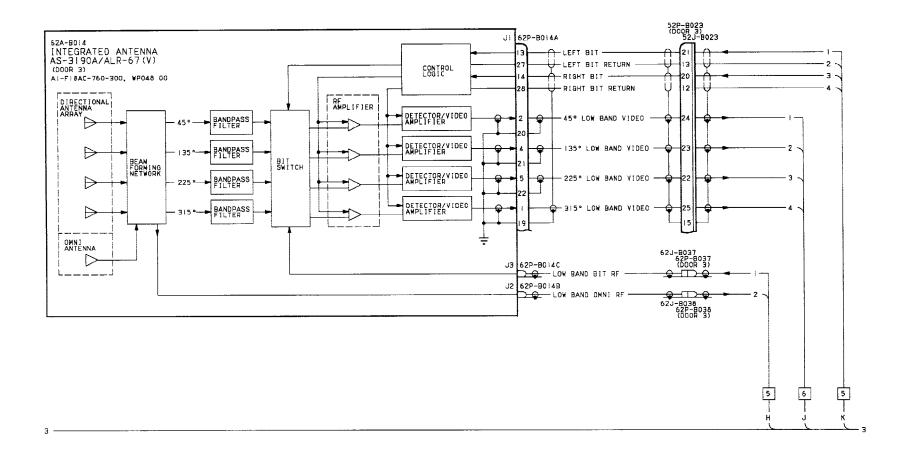


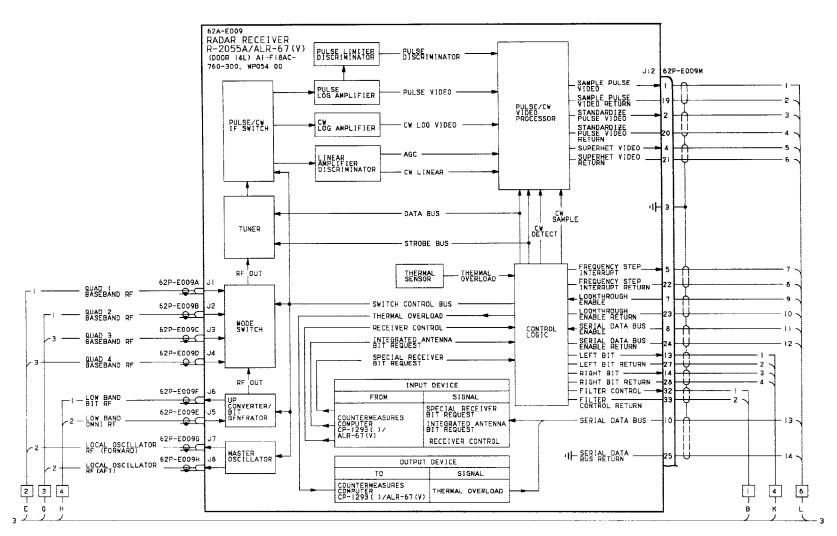


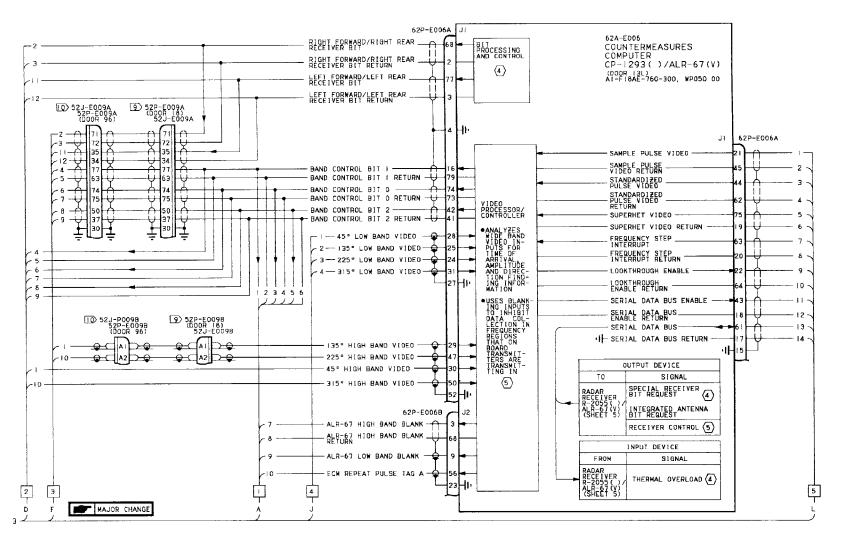
Change 1



18AC-760-50-(11-3)16-GRID

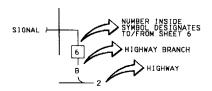






LEGEND

I. NONSTANDARD SYMBOLS:



- \bigoplus identifies relay used to switch low level current. See Note 2.
- 2. CONTINUITY TESTS:
 - A. ALL AIRCRAFT WIRE NUMBERS, SPLICE POINTS, AND GROUND POINTS ARE SHOWN IN AI-FIBA()-WDM-000.
 - B. MIER A LOW LEYEL CURRENT SMITCHING RELAY (IDENTIFIED BY B) IS REMOVED FOR TROUBLESHOOTING, IDENTIFY RELAY AND SOCKET FOR CORRECT REINSTALLATION DO NOT REPLACE LOW LEVEL CURRENT SMITCHING RELAY WITH ANY OTHER USED RELAY. IF RELAY IS DEFECTIVE, REPLACE WITH NEW RELAY.
 - C. DO NOT TEST LOW LEVEL DEVICES (SMITCHES/RELAY CONTACTS)
 FOR CONTINUITY WITH MULTIMETER ON RXI SCALE PIN TO
 PIN TESTS THAT DO NOT GO THROUGH SWITCHES/RELAY CONTACTS
 MAY USE THE RXI SCALE.
 - D. WHEN TESTING CONTINUITY TEST FOR:
 - (I) SHORTS TO GROUND.
 - (2) SHORTS BETWEEN SURROUNDING PINS ON CONNECTORS.
 - (3) SHORTS BETWEEN SHIELD AND CONDUCTORS.
 - (4) SHIELD CONTINUITY.
- 3. LINE UNDER LETTER (S) INDICATES LOWER CASE PIN LETTER.
- (4) CONTROLS, DISPLAYS AND AUDIO SCHEMATIC, WPD15 DO.
- (5) VIDEO PROCESSING AND CONTROL SCHEMATIC, WP014 00.
- (6) INTERFERENCE BLANKER FUNCTIONAL SCHEMATIC, WP004 00.
- (7) DC POWER SYSTEM SCHEMATIC, AI-FIBAC-420-500, WP004 00.
- (8) GROUND POWER SWITCHING SCHEMATIC, AI-FIBAC-420-500, WP005 00
- 9 F/A-18A
- 10 F/A-18B

1 September 1992

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

SCHEMATIC - INTEGRATION

COUNTERMEASURES WARNING AND CONTROL SYSTEM

EFFECTIVITY: 161702 AND UP

Reference Material

None

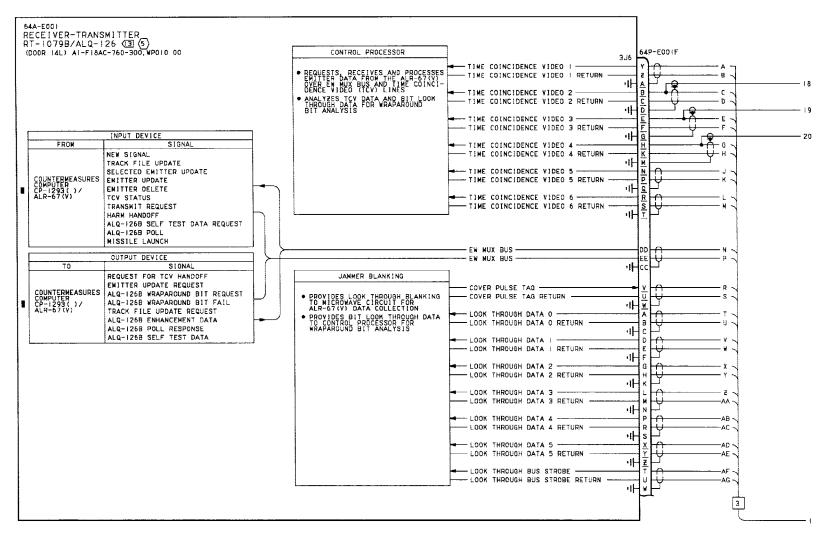
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Record of Applicable Technical Directives

None

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Page 2



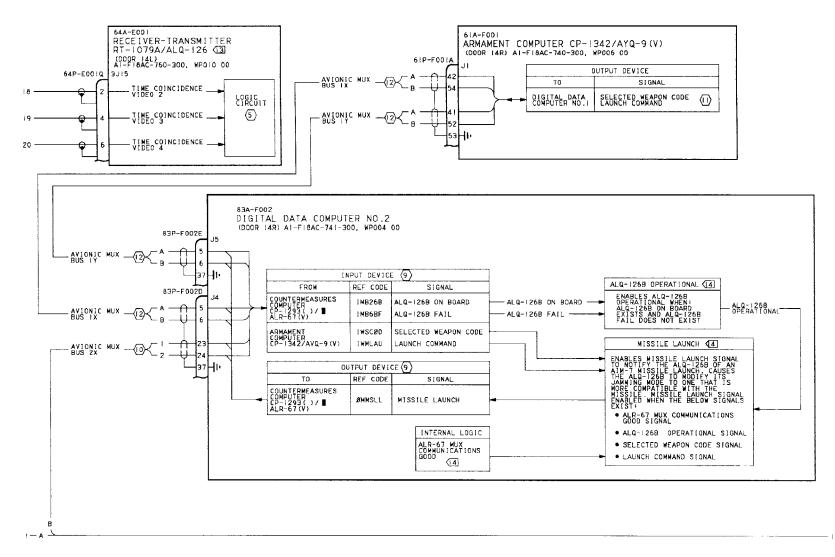
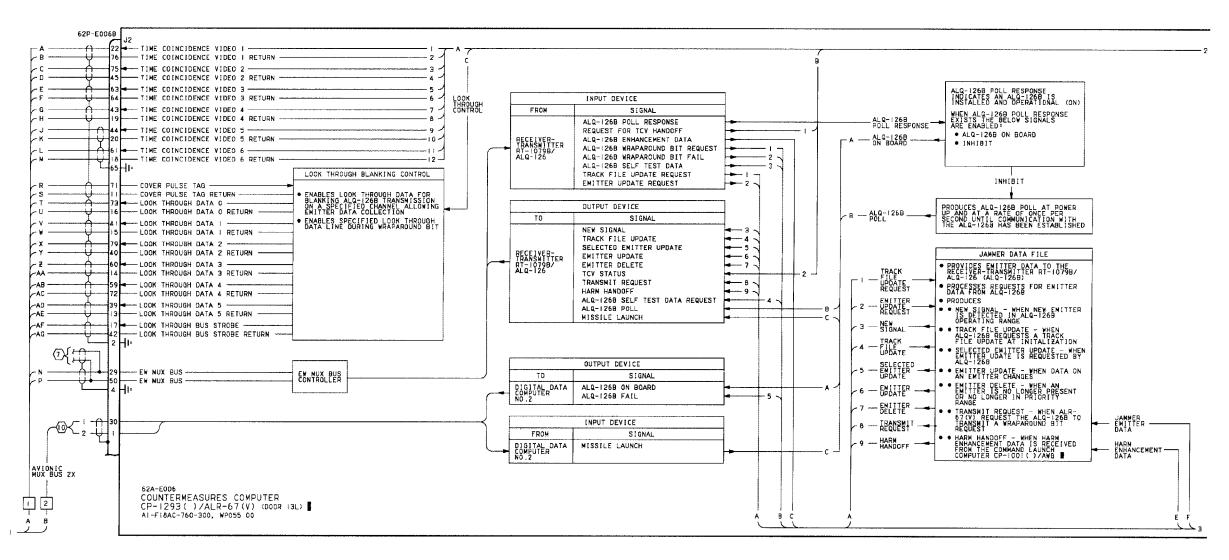
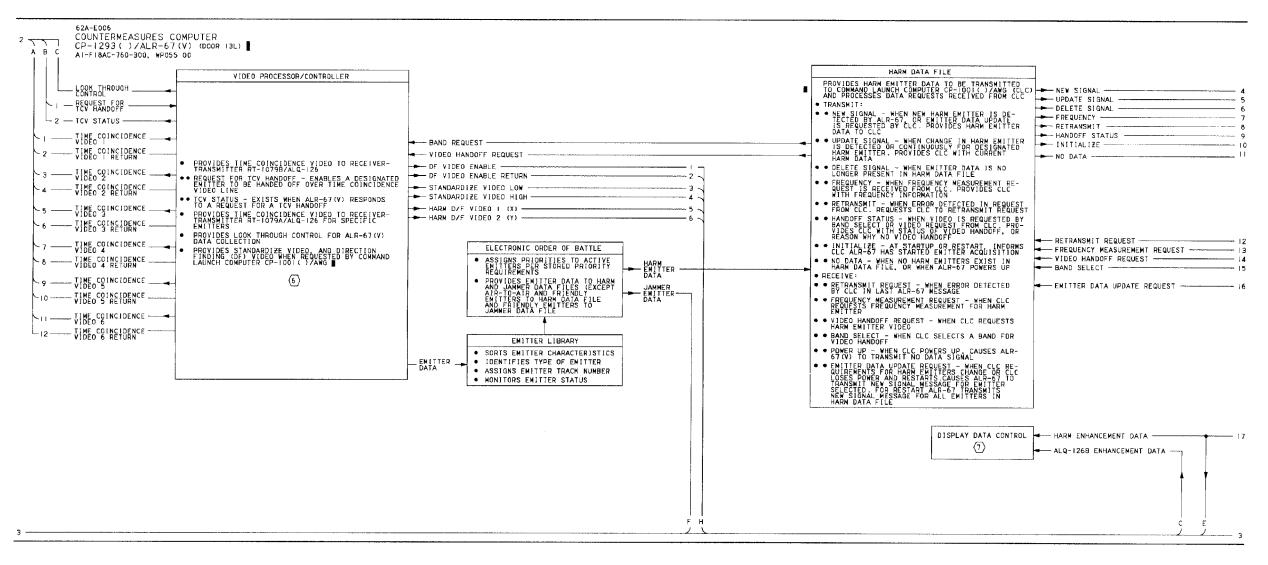
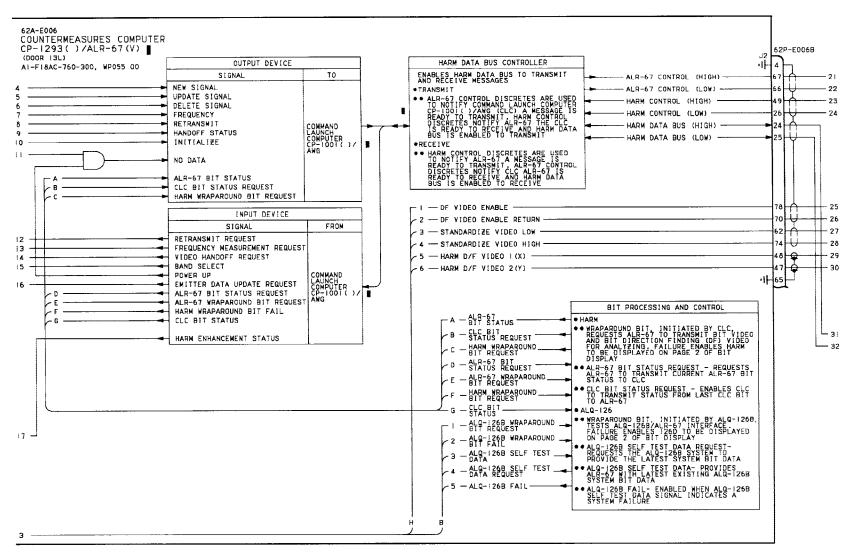


Figure 1.

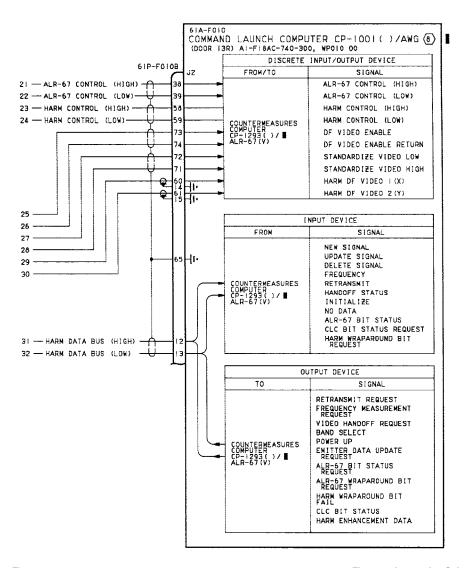






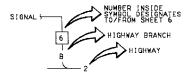
18AC-760-50-(13-5)16-GRID

A1-F18AC-760-500 013 00 Page 7/(8 blank)



LEGEND

I. NONSTANDARD SYMBOLS:



- 2. CONTINUITY TEST:
 - A. ALL AIRCRAFT WIRE NUMBERS, SPLICE POINTS AND GROUND POINTS ARE SHOWN IN AI-F18A()-WDM-000.
 - B. WHEN A LOW LEVEL CURRENT SWITCHING RELAY (IDENTIFIED BY \bigoplus) IS REMOVED FOR TROUBLESHOOTING, IDENTIFY RELAY AND SOCKET FOR CORRECT REINSTALLATION, DO NOT REPLACE LOW LEVEL CURRENT SWITCHING RELAY MITH, ANY OTHER USED RELAY, IF RELAY IS DEFECTIVE, REPLACE WITH NEW
 - C. DO NOT TEST LOW LEVEL DEVICES (SWITCHES/RELAY CONTACTS) FOR CONTINUITY WITH MULTIPHETER ON RXI SCALE, PIN TO PIN TESTS THAT DO NOT GO HARDOUGH SWITCHES/RELAY CONTACTS MAY USE THE RXI SCALE.
 - D. WHEN TESTING CONTINUITY, TEST FOR:
 - (1) SHORTS TO GROUND STROUNDING PINS ON CONNECTORS. (2) SHORTS BETWEEN SHIELD AND CONNECTORS. (3) SHORTS BETWEEN SHIELD AND CONNECTORS. (4) SHIELD CONTINUITY.
- 3. LINE UNDER LETTER (S) INDICATES LOWER CASE PIN LETTERS.
- 4. NONSTANDARD ABBREVIATIONS:

BIT - BUILT-IN TEST. HARM - HIGHSPEED ANII-RADIATION MISSILE. ALR-67 - COUNTERMEASURES WARNING AND CONTROL SYSTEM.

- (5) RECEIVER TRANSMITTER RT-1079B/ALQ-126 SCHEMATIC, WP008 00.
- (6) VIDEO PROCESSING AND CONTROL SCHEMATIC, WP014 00.
- (7) CONTROLS, DISPLAYS AND AUDIO SCHEMATIC, WP015 00.
- (8) AGM-68 HARM AVIONIC INTERFACE SCHEMATIC, AI-F18AC-740-500, WP059 00.
- (9) FOR LUGIC PLAGRAMS RELATING TO REF CODE, REFER TO AL-FLBA()-DLD-000 AL-FLBAC-FLM-100.
- (10) AVIONIC MUX CHANNEL 2 SCHEMATIC, AI-FIBAC-741-500, WP005 00.
- (I) AIM-7 SPARROW AVIONIC INTERFACE SCHEMATIC, AI-FIBAC-740-500, WP041 00.
- (2) AVIONIC MUX CHANNEL I SCHEMATIC, AI-FIBAC-741-500, WPO04 00
- RECEIVER-TRANSMITTER RT-1079A/ALQ-126 AND RECEIVER-TRANSMITTER RT-1079B/ALQ-126 ARE ALTERNATE CONFIGURATIONS OF COUNTERMEASURES SET.
- WITH DIGITAL DATA COMPUTER CONFIG/IDENT NO.84A AND UP

18AC-760-50-(13-6)16-GRID

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

SCHEMATIC - VIDEO PROCESSING AND CONTROL

COUNTERMEASURES WARNING AND CONTROL SYSTEM

EFFECTIVITY:161702 AND UP

This WP supersedes WP014 00, dated 1 September 1992.

Reference Material

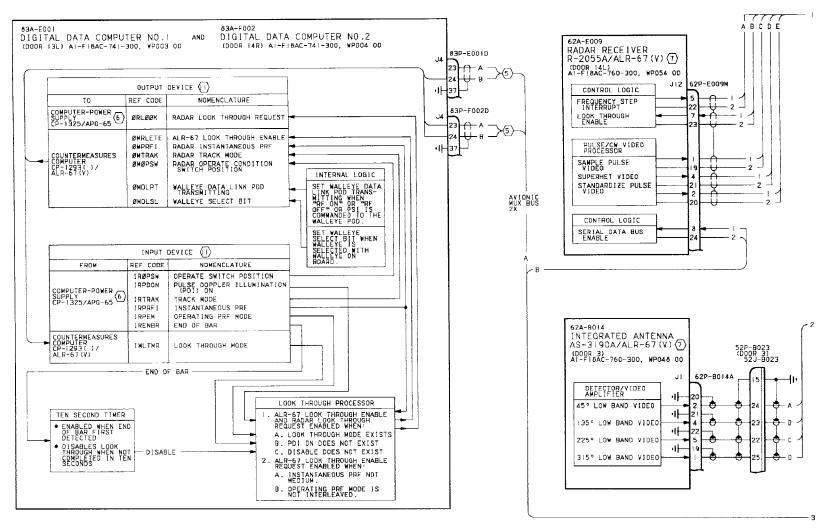
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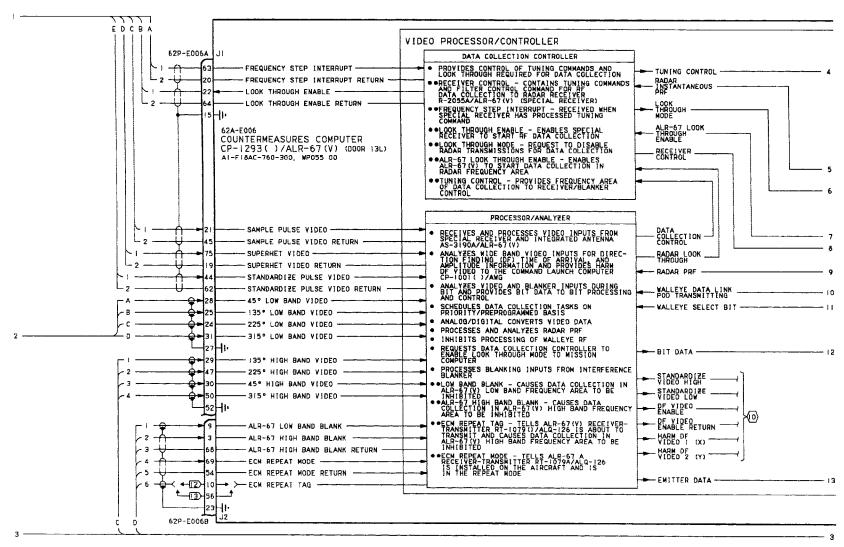
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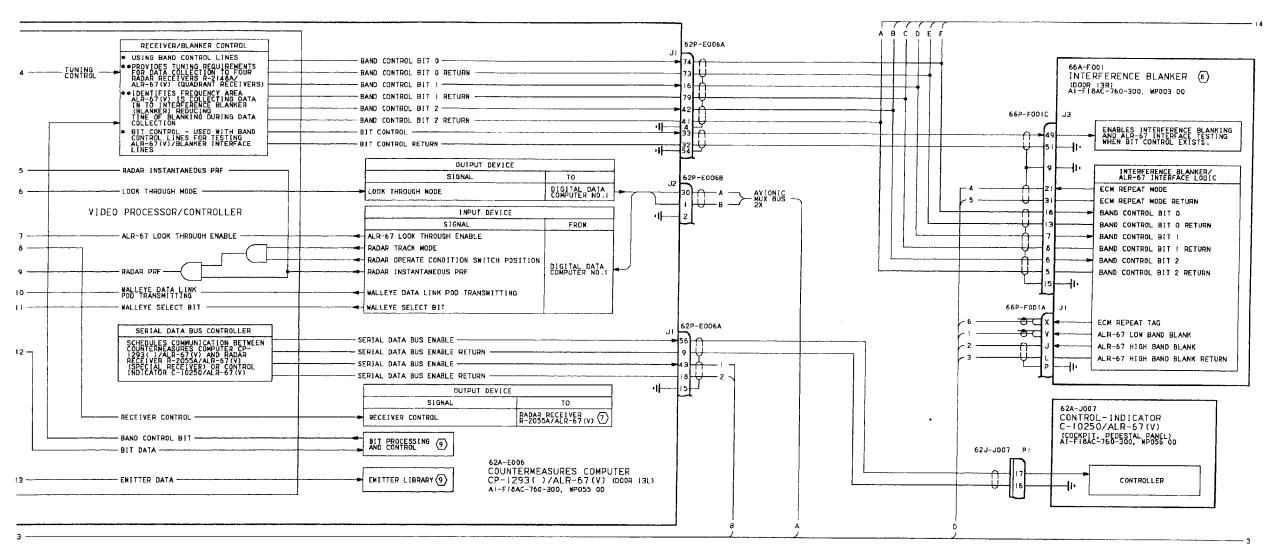
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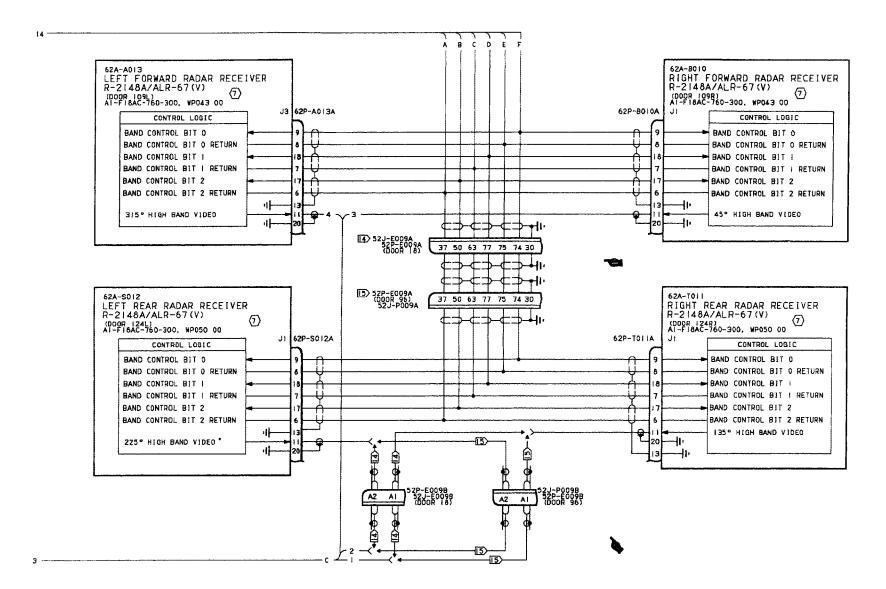
Record of Applicable Technical Directives

Type/ Number	Date	Title and ECP No.	Date Incorp.	Remarks
F/A-18 AFC 158	Oct 91	Correction of AN/ALR-67 and AN/ ALQ-126B Wiring (ECP RAMEC NO- RIS-22-90)	1 Sep 92	-



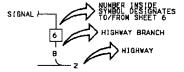






LEGEND

- I. NONSTANDARD SYMBOLS:
- IDENTIFIES RELAY USED TO SWITCH LOW LEVEL CURRENT.
 SEE NOTE 2.



- 2. CONTINUITY TESTS:
- A. ALL AIRCRAFT WIRE NUMBERS, SPLICE POINTS AND GROUND POINTS ARE SHOWN IN AI-FIBA()-WDM-000.
- C. DO NOT TEST LOW LEVEL DEVICES (SWITCHES/RELAY WITH MULTI METER ON RXI SCALE, PIN TO PIN TESTS NOT GO THROUGH SWITCHES/RELAY CONTACTS MAY USE SCALE.
- D. WHEN TESTING CONTINUITY, TEST FOR:
 - (I) SHORTS TO GROUND.
 - (2) SHORTS BETWEEN SURROUNDING PINS ON CONNECTORS.
 - (3) SHORTS BETWEEN SHIELD AND CONNECTORS
- (4) SHIELD CONTINUITY.
- 3. LINE UNDER LETTER (S) INDICATES LOWER CASE PIN LETTER.
- 4. NONSTANDARD ABBREVIATIONS

BIT - BUILT-IN TEST

HARM - HIGHSPEED ANTI-RADIATION MISSILE ALR-67(V) - COUNTERMEASURES WARNING AND CONTROL SYSTEM

- (5) AVIONIC MUX CHANNEL 2 SCHEMATIC AI-FIBAC-741-500, WP005 00.
- TWS TARGETS AND LAUNCH RANGE AND STEERING TARGETS SCHEMATIC, A1-F18AC-742-500, WOOZI DO.
- (7) RF DETECTION AND CONVERSION SCHEMATIC, WP012 00.
- (8) INTERFERENCE BLANKER SYSTEM FUNCTIONAL SCHEMATIC, WP004 00.
- (9) CONTROLS, DISPLAYS AND AUDIO SCHEMATIG, WPO15 00.
- (10) INTEGRATION SCHEMATIC, WP013 00.
- FOR LOGIC DIAGRAMS RELATING TO REFERENCE ACRESEN TO ATTION RELATION REFERENCE FOR THE PROPERTY OF THE PROPERTY
- [2] 161702 THRU 161736 BEFORE F/A-18A AFC 158.
- [3] 161737 AND UP: ALSO 161702 THRU 161736 AFTER F/A-18A AFC 158.
- [4] F/A-18A
- [5] F/A-18B

1 September 1992

ORGANIZATIONAL MAINTENANCE

SYSTEM SCHEMATICS

SCHEMATIC - CONTROLS, DISPLAYS AND AUDIO

COUNTERMEASURES WARNING AND CONTROL SYSTEM

EFFECTIVITY: 161702 AND UP

Reference Material

None

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Record of Applicable Technical Directives

Type/ Number	Date	Title and ECP No.	Date Incorp.	Remarks
F/A-18 AFC 50	16 Oct 84	Tactical Electronic Warfare Systems, ALR-67 Countermeasures, Modification of (ECP MDA-F/A-18-003R1 C1/C2/C3)	1 Oct 84	-

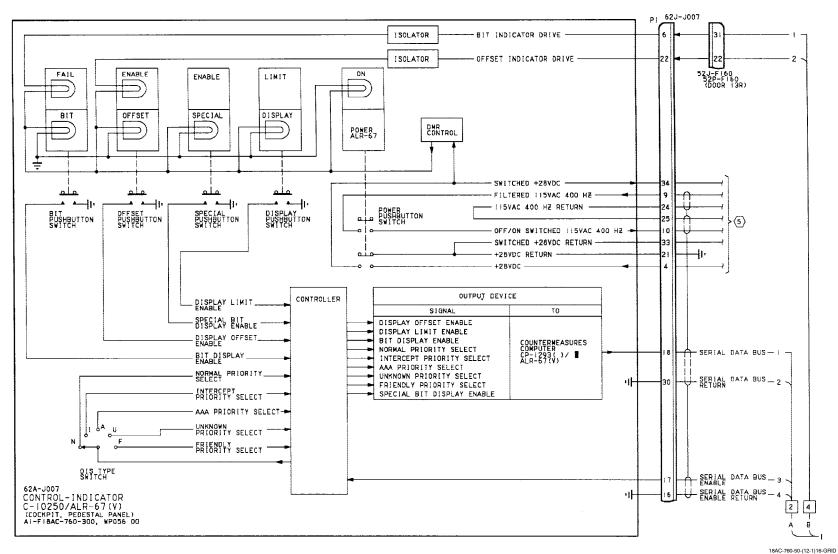
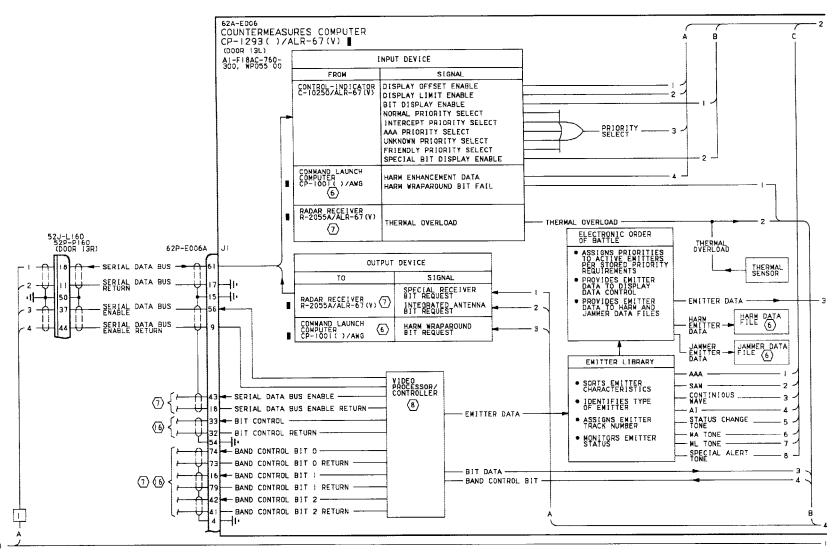


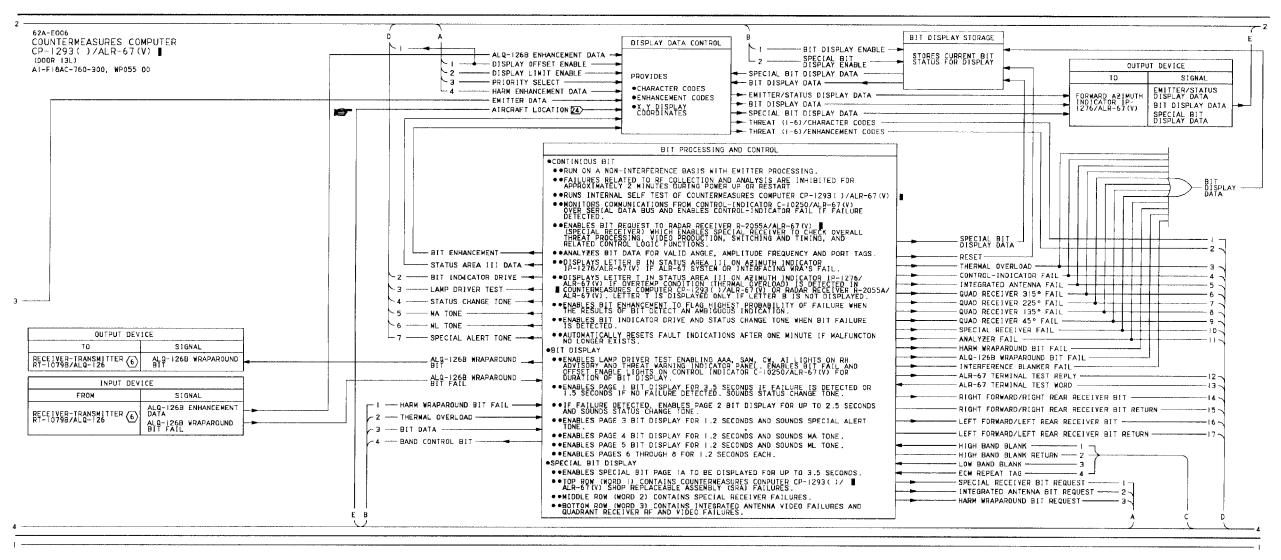
Figure 1.

Figure 1. Controls Displays and Audio Schematic (Sheet 1)



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Figure 1. Controls Displays and Audio Schematic (Sheet 2)



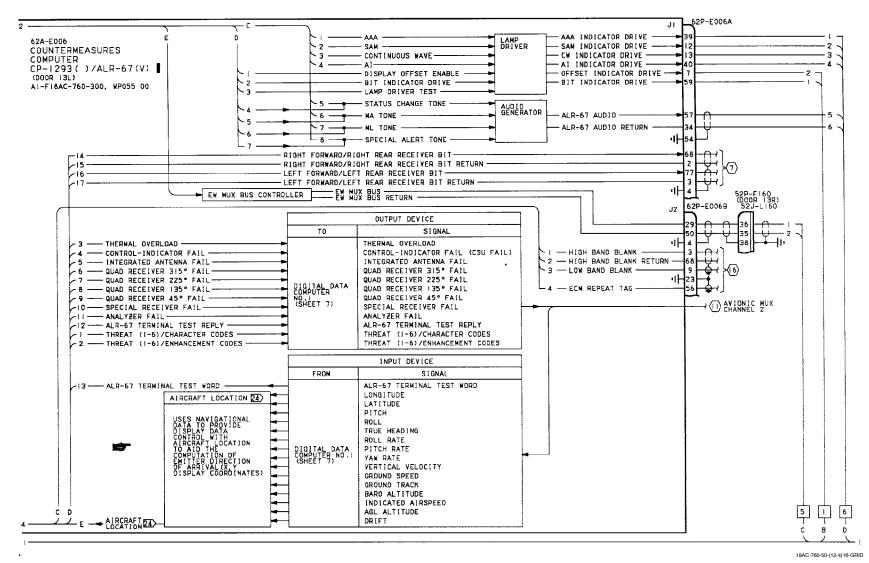
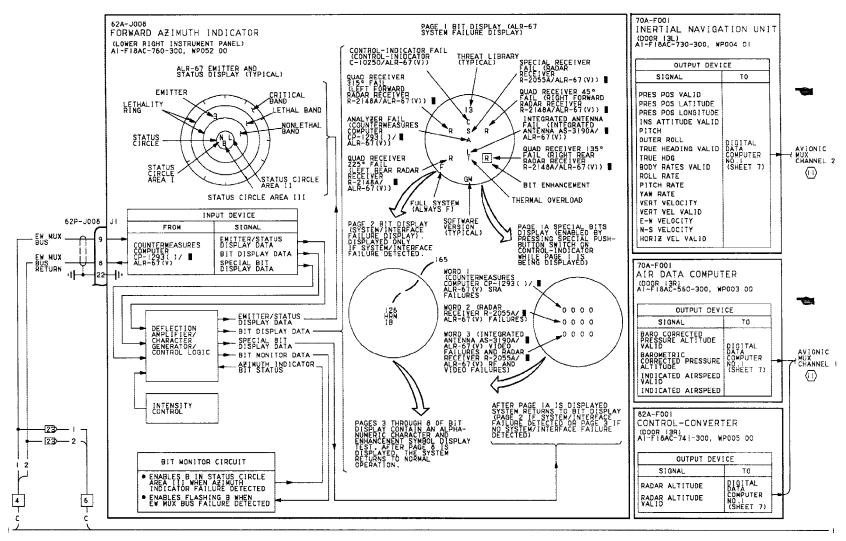


Figure 1. Controls Displays and Audio Schematic (Sheet 4)

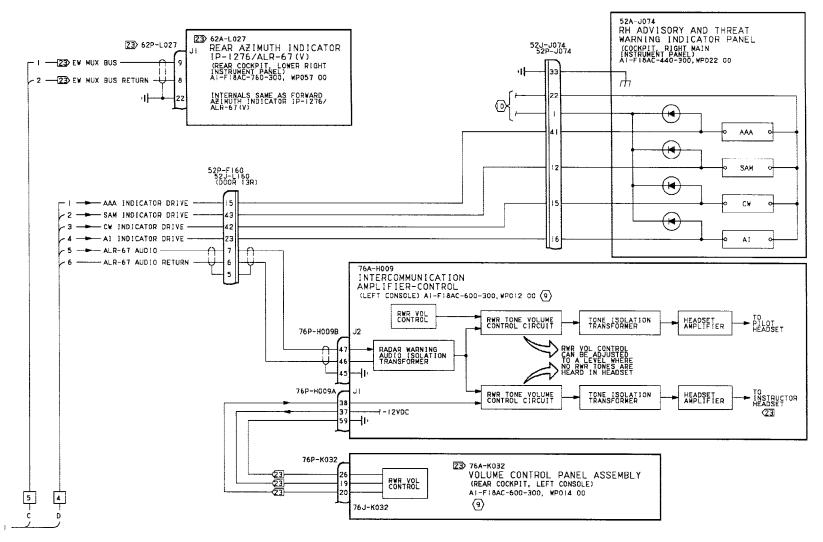
Figure 1.

A1-F18AC-760-500

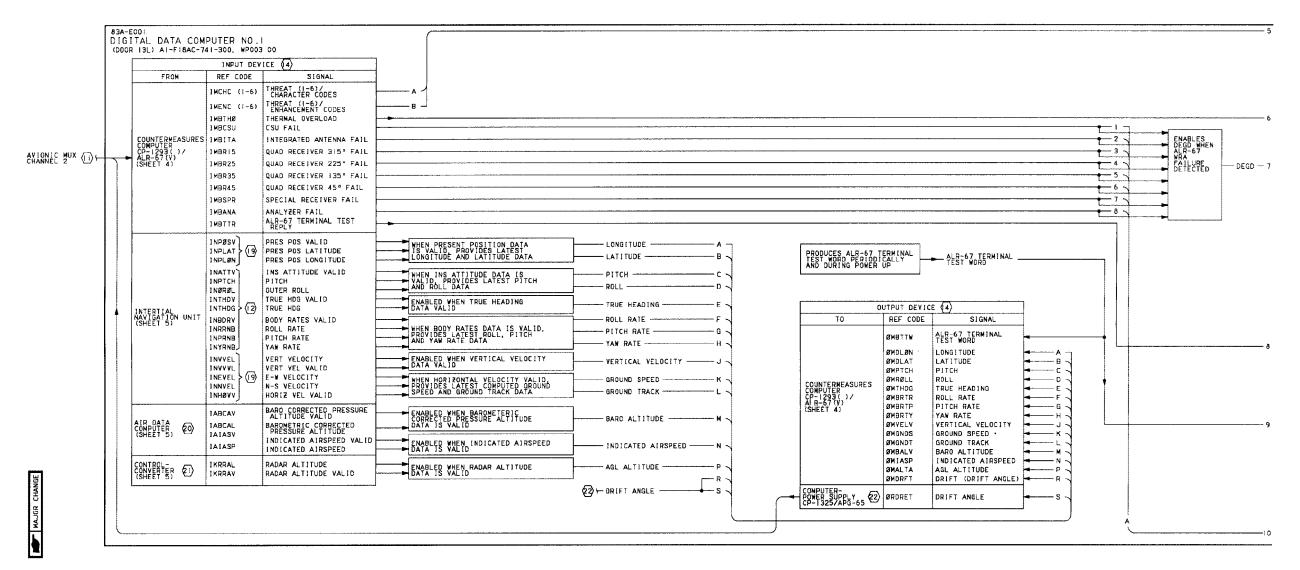


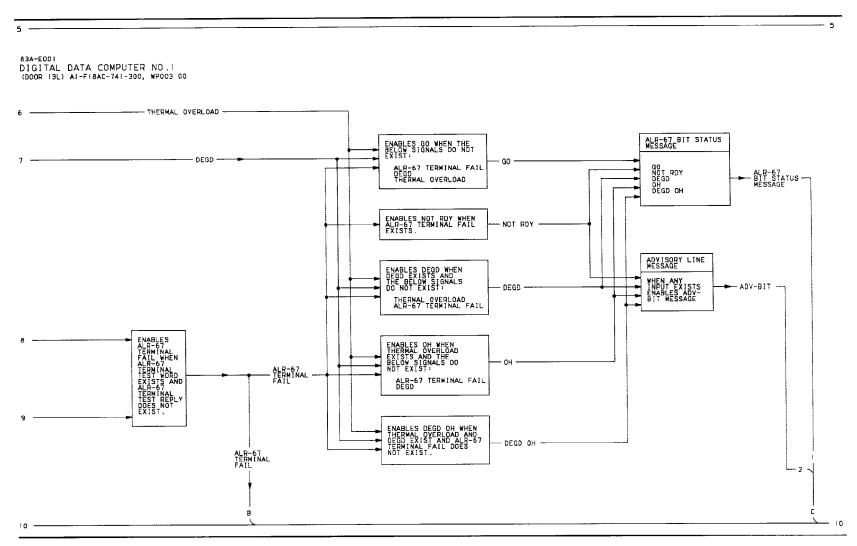
18AC-760-50-(12-5)16-GRID

A1-F18AC-760-500 015 00
Page 7

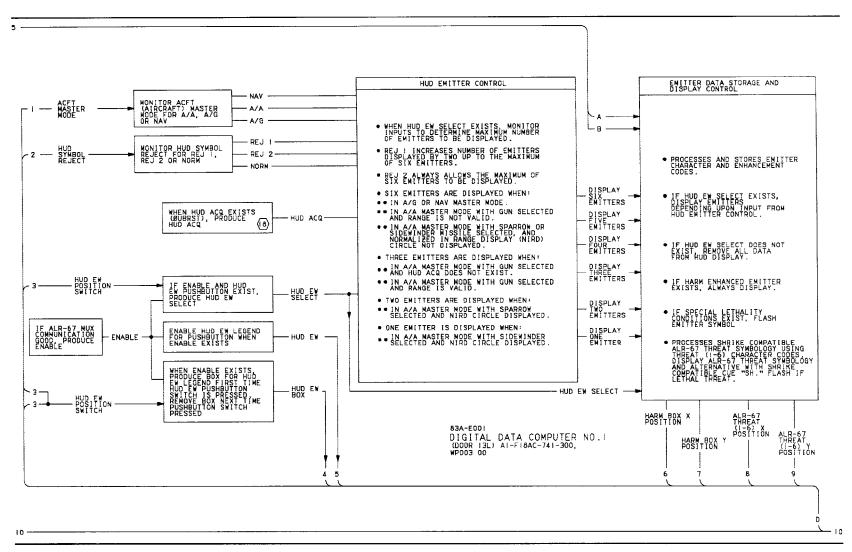


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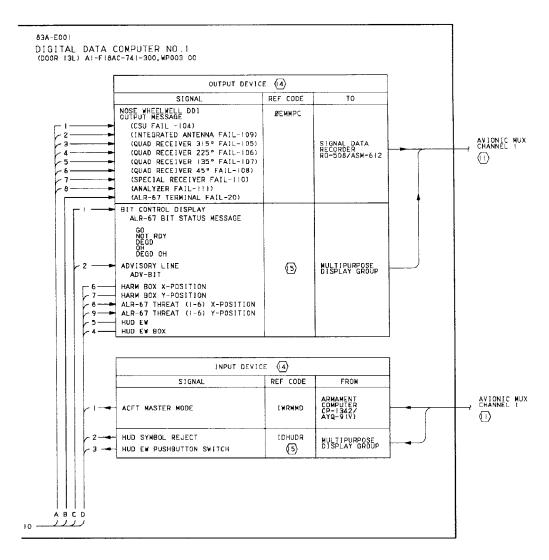


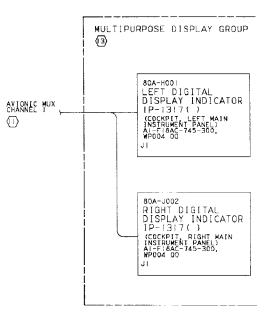


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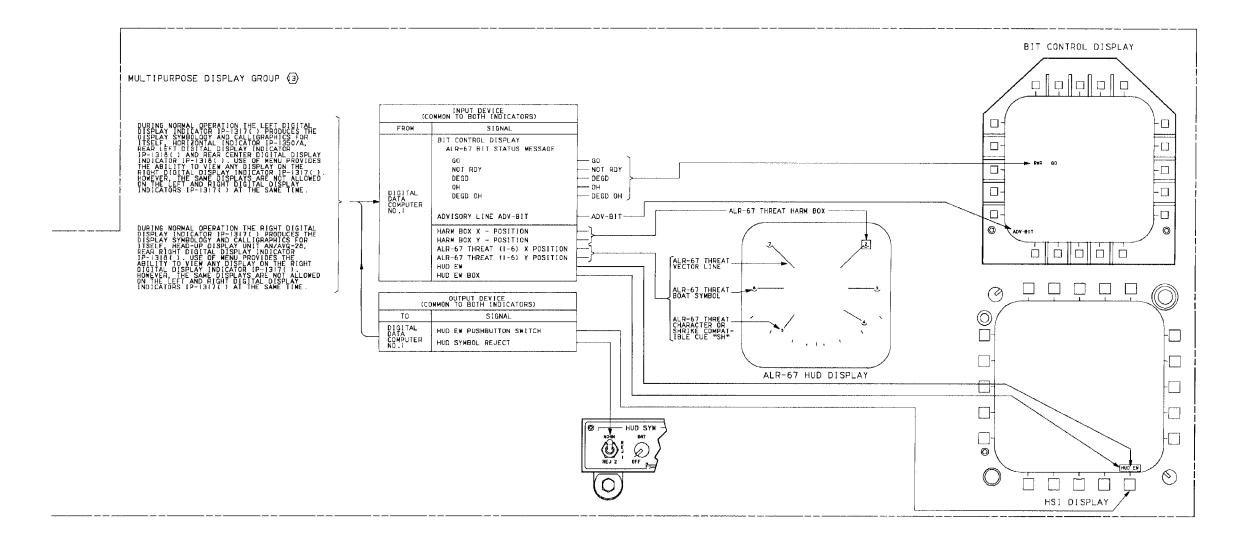
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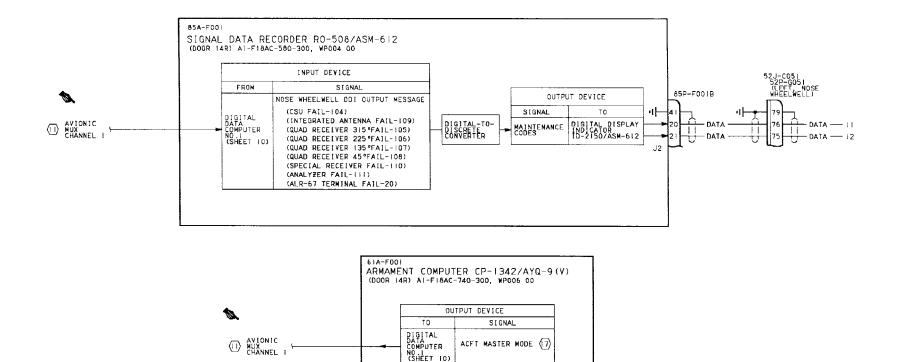


18AC-760-50-(12-10)16-GRID

Figure 1. Controls Displays and Audio Schematic (Sheet 10)

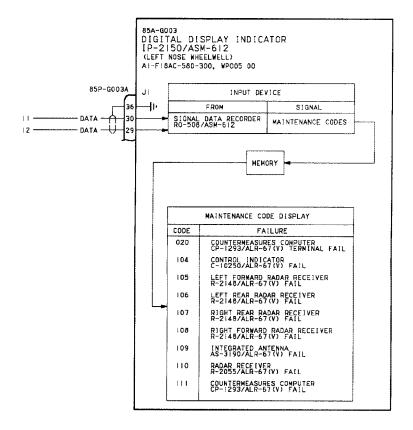


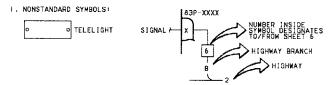
A1-F18AC-760-500 015 00
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A1-F18AC-760-500







IDENTIFIES RELAY USED TO SWITCH LOW LEVEL CURRENT. SEE NOTE 2.

- A. ALL AIRCRAFT WIRE NUMBERS, SPLICE POINTS, AND GROUND POINTS ARE SHOWN IN AI-F18A()-WDM-000.
- B. WHEN A LOW LEVEL CURRENT SWITCHING RELAY (IDENTIFIED BY \bigoplus) IS REMOVED FOR TROUBLESHOOTING, IDENTIFY RELAY AND SOCKET FOR CORRECT REINSTALLATION. DO NOT REPLACE LOW LEVEL CURRENT SWITCHING RELAY MITH ANY OTHER USED RELAY. IF RELAY IS DEFECTIVE, REPLACE WITH NEW RELAY.
- C. DO NOT TEST LOW LEVEL DEVICES (SWITCHES/RELAY CONTACTS) FOR CONTINUITY SWITCHES/RELAY CONTACTS SCALE THAT DO NOT GO THROUGH
- D. WHEN TESTING CONTINUITY, TEST FOR
 - (I) SHORTS TO GROUND.
 - (2) SHORTS BETWEEN SURROUNDING PINS ON CONNECTORS.
 - (3) SHORTS BETWEEN SHIELD AND CONDUCTORS.
 - (4) SHIELD CONTINUITY.
- 3. LINE UNDER LETTER (S) INDICATES LOWER CASE PIN LETTER.
- 4. NONSTANDARD ABBREVIATIONS:
- BIT BUILT-IN TEST
- HARM HIGHSPEED ANTI-RADIATION MISSILE
- EW ELECTRONIC WARFARE
- MA MISSILE ALERT
- ML MISSILE LAUNCH
- (5) POWER INTERFACE SCHEMATIC, WPOIL OO.
- (6) INTEGRATION SCHEMATIC, WP013 00.
- (7) RF DETECTION AND CONVERSION SCHEMATIC. WPD12 GO.
- (8) VIDEO PROCESSING AND CONTROL SCHEMATIC, WPO14 00.
- (9) INTERCOMMUNICATION AND AUDIO SYSTEM FUNCTIONAL SCHEMATIC AI-FIBAC-600-500,

- (0) COCKPIT ADVISORY LIGHTS SCHEMATIC AI-FIBAC-440-500, WP006 00
- (T) SEE APPLICABLE AVIONIC MUX CHANNEL SCHEMATIC AI-F18AC-741-500.
- (2) NAVIGATION ATTITUDE AND HEADING FUNCTIONAL SCHEMATIC, AI-FIBAC-730-500, MPDI7 00.

- (5) BISPLAY REF CODES AFE DOTATSHOWN F MALFUNCTION MAY FUNCTION MEXISTEAN TRANSFER INDICATOR. TROUBLESHOOT USING A 1-F18AC-01D-000 IMPUT REF CODES F MALFUNCTION EXISTS ON MEXIST ON THE CODE OF MALFUNCTION EXISTS ON MEXISTS OF MALFUNCTION EXISTS ON MEXIST OF MALFUNCTION EXISTS OF MEXISTS OF MEXIST
- (6) INTERFERENCE BLANKER FUNCTIONAL SCHEMATIC, WP004 00.
- (7) AIRCRAFT MASTER MODE SELECT SCHEMATIC, AI-FIBAC-740-500, WPOI3 00.
- (8) AIR COMBAT MANEUVERING AND GUN MODE DISPLAY SCHEMATIC, AI-F:8AC-742-500, MP022 00.
- (19) ANY IGATION VELOCITY AND POSITION KEEPING FUNCTIONAL SCHEMATIC, AND FOREST
- (20) AIR DATA COMPUTER SYSTEM FUNCTIONAL SCHEMATIC, AI-FIBAC-560-500, MP004 00.
- DELECTRONIC ALTIMETER SYSTEM FUNCTIONAL SCHEMATIC, AI-FIBAC-600-500,
- | ② AIR TO GROUND ANTENNA CONTROL FUNCTIONAL SCHEMATIC, A1-F18AC-742-500,
- 23 F/A-18B
- 24 WITH DIGITAL DATA COMPUTER NO.: CONFIG/IDENT NO.92A AND UP